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- 18 Havener, A.G. and Radley, R.J. Supersonic wind tunnel investigation using pulsed laser holography. ARL 73-0148 (1973)
- 19 Mabey, D.G. and Gaudet, L. Some applications of small skin friction balances at supersonic speeds. *AIAA Journal of Aircraft*, Vol. 12, No. 10, pp 819-825 (1975)
- 20 Lee, R.E., Yanta, W.J. and A.C. Leonas Velocity profile, skin friction balance and heat transfer measurements of the turbulent boundary layer at Mach 5 and zero pressure gradient. NOL TR69 106, June 1969
- 21 Voisinet, R.L.P. and Lee, R.E. Measurements of a Mach 4.9 zero pressure gradient turbulent boundary layer with heat transfer Part I - Data compilation. NOL TR72-232 (1972)
- 22 Lee, R.E. and Voisinet, R.L.P. Contribution to Euromech 43 Colloquium. Gottingen, May 1973
- 23 Bushnell, D.M., Cary, A.M. and Holley, B.B. Mixing length in low Reynolds number compressible turbulent boundary layers. *AIAA Journal*, Vol. 13, 8 p 1119 to 1121, August 1975

## CORRIGENDUM

### Pressure Gradient and Leading-Edge Effects on the Corner Boundary Layer

by

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The identifying symbols are missing from the keys below Figs. 2 and 3. The key for Fig. 3 also applies to subsequent Figures.

They should read as follows:-

Fig. 2

■	$\alpha \approx 6^\circ$		sharp nosed L.E.	$\alpha \approx 0^\circ$	round nosed L.E.
○	$\alpha \approx 1^\circ$				
●	$\alpha \approx 0.2^\circ$				

Fig. 3

$Z_o/\sqrt{2}$	=	0.5mm ○	1.4mm ●	2.8mm ▲	5.0mm ▲
		7.8mm ▽	11.4mm ▼	18.0mm □	36.0mm ■