

Asymptotics of the centre-mode instability in viscoelastic channel flow: with and without inertia

Rich R. Kerswell^{1,†} and Jacob Page^{2,†}

¹DAMTP, Centre for Mathematical Sciences, University of Cambridge, CB3 0WA, UK

²School of Mathematics, University of Edinburgh, EH9 3FD, UK

(Received 14 December 2023; revised 24 April 2024; accepted 14 May 2024)

Motivated by the recent numerical results of Khalid *et al.* (*Phys. Rev. Lett.*, vol. 127, 2021, 134502), we consider the large-Weissenberg-number (W) asymptotics of the centre mode instability in inertialess viscoelastic channel flow. The instability is of the critical layer type in the distinguished ultra-dilute limit where $W(1 - \beta) = O(1)$ as $W \rightarrow \infty$ (β is the ratio of solvent-to-total viscosity). In contrast to centre modes in the Orr–Sommerfeld equation, $1 - c = O(1)$ as $W \rightarrow \infty$, where c is the phase speed normalised by the centreline speed as a central ‘outer’ region is always needed to adjust the non-zero cross-stream velocity at the critical layer down to zero at the centreline. The critical layer acts as a pair of intense ‘bellows’ which blows the flow streamlines apart locally and then sucks them back together again. This compression/rarefaction amplifies the streamwise-normal polymer stress which in turn drives the streamwise flow through local polymer stresses at the critical layer. The streamwise flow energises the cross-stream flow via continuity which in turn intensifies the critical layer to close the cycle. We also treat the large-Reynolds-number (Re) asymptotic structure of the upper (where $1 - c = O(Re^{-2/3})$) and lower branches of the Re – W neutral curve, confirming the inferred scalings from previous numerical computations. Finally, we remark that the viscoelastic centre-mode instability was actually first observed in viscoelastic Kolmogorov flow by Boffetta *et al.* (*J. Fluid Mech.*, vol. 523, 2005, pp. 161–170).

Key words: polymers, viscoelasticity

[†] Email addresses for correspondence: rrk26@cam.ac.uk, jacob.page@ed.ac.uk

1. Introduction

It is well known that the addition of long-chain polymers to a Newtonian fluid introduces elasticity which can give rise to fascinating new ‘viscoelastic’ flow phenomena. Prime examples of this are a new form of spatio-temporal chaos – dubbed ‘elastic’ turbulence (ET) (Groisman & Steinberg 2000) – which exists in inertialess curvilinear flows and ‘elasto-inertial’ turbulence (EIT) (Samanta *et al.* 2013) which can occur in two-dimensional rectilinear flows (Sid, Terrapon & Dubief 2018) where inertia and elasticity balance each other. While ET is assumed triggered by a linear ‘hoop stress’ instability of curved streamlines (Larson, Shaqfeh & Muller 1990; Shaqfeh 1996), the origin of EIT remains unclear (Datta *et al.* 2022; Dubief, Terrapon & Hof 2023) as does any possible relationship to ET.

The breakdown of viscoelastically modified Tollmien–Schlichting modes has been suggested as a cause of EIT (Shekar *et al.* 2019, 2021) at least at high Reynolds number, Re , and low Weissenberg number, W . At low Re and high W , however, the recent discovery of a new linear instability of rectilinear viscoelastic shear flow seems more viable (Garg *et al.* 2018; Chaudhary *et al.* 2021; Khalid *et al.* 2021a). This instability occurs at higher W than generally associated with EIT but has been shown to be subcritical (Page, Dubief & Kerswell 2020; Wan, Sun & Zhang 2021; Buza, Page & Kerswell 2022b; Buza *et al.* 2022a). In particular, travelling wave solutions, which have a distinctive ‘arrowhead’ structure, originating from the neutral curve reach down in W to where EIT exists in parameter space (Page *et al.* 2020; Buza *et al.* 2022a; Dubief *et al.* 2022). This instability is of centre-mode type, being localised either at the centre of a pipe (Garg *et al.* 2018; Chaudhary *et al.* 2021) or midplane of a channel (Khalid *et al.* 2021a; Khalid, Shankar & Subramanian 2021b), but is notably absent in plane-Couette flow (Garg *et al.* 2018). Perhaps most intriguingly, the instability can be traced down to $Re = 0$ in channel flow (Khalid *et al.* 2021b) in the ultra-dilute limit of the solvent-to-total viscosity ratio approaching 1 while a minimum $Re \approx 63$ exists in pipe flow (Chaudhary *et al.* 2021). Subsequently, travelling wave solutions have been numerically computed in two dimensions and at $Re = 0$ (Buza *et al.* 2022a; Morozov 2022) and their instability examined (Lellep, Linkmann & Morozov 2023, 2024).

Apart from numerically inferred scaling relationships (Garg *et al.* 2018; Chaudhary *et al.* 2021; Khalid *et al.* 2021a,b), the only work to unpick the asymptotic structure of the centre-mode instability is that of Dong & Zhang (2022) in pipe flow. They identify the asymptotic structure on the upper branch of the neutral curve characterised by $W \sim Re^{1/3}$ as $Re \rightarrow \infty$ and consider the long-wavelength limit but stop short of treating the lower branch of the neutral curve. Here, we do both for the channel and go further to examine the inertialess regime in channel flow which is absent in pipe flow. Unravelling the $Re = 0$ situation asymptotically is actually our main motivation here as it differs fundamentally from all the classical Orr–Sommerfeld work performed for Newtonian shear flows (Drazin & Reid 1981). In particular, the regularising feature of the critical layer formed (e.g. figure 3 of Khalid *et al.* 2021b and figure 4 below) is the presence of elastic relaxation rather than viscosity. The ‘outer’ relaxation-free solutions also satisfy a fourth-order differential equation rather than the classical, inviscid, second-order Rayleigh equation in Newtonian flows. This means that matching conditions across the critical layer need to be sought down to the third-order derivative in the cross-stream velocity (or streamfunction) and, due to a logarithmic singularity in the first-order derivative, computations need to go beyond double precision accuracy to achieve a convincing correspondence between numerical results and the asymptotic predictions; see table 4. A particularly interesting feature of this viscoelastic centre-mode instability is that the critical layer does not

approach the midplane as $W \rightarrow \infty$, that is, the phase speed of the instability approaches a non-trivial value very close to but distinct from 1, the maximum speed of the base flow. Ultimately, however, the point of the asymptotic analysis is to identify the mechanism of the instability and to understand, if possible, why it does not manifest in plane-Couette flow.

The plan of this paper is first to introduce the channel flow problem in § 2 and the viscoelastic model (Oldroyd-B) used by Khalid *et al.* (2021a,b). The first results section, § 3, then examines the large Re -asymptotics of the upper (§ 3.1) and lower branches (§ 3.2) of the neutral curve in the Re - W plane for fixed β , the ratio of solvent-to-total viscosity: see figure 1. Reduced eigenvalue problems based only on $O(1)$ quantities (relative to Re) can be straightforwardly derived for both upper and lower branches. Interestingly, if $\beta \gtrsim 0.9905$, Khalid *et al.* (2021b) showed that the lower branch crosses the $Re = 0$ axis and the appropriate (mathematical) limit is then $Re \rightarrow -\infty$. Figure 3 indicates that nothing mathematically unusual happens as the neutral curve swings around from pointing at $Re \rightarrow \infty$ to $Re \rightarrow -\infty$ although, of course, negative Re makes little physical sense. The special case of $Re = 0$ or vanishing inertia, however, does and the asymptotics as $W \rightarrow \infty$ is studied in § 4. The work of Khalid *et al.* (2021b) has already indicated that the appropriate distinguished limit is that in which β simultaneously approaches 1 such that $W(1 - \beta)$ stays finite. Section 5 goes on to use the asymptotic solution to discuss the mechanics of the inertialess instability and § 6 describes some numerical experiments to understand how the instability responds to the problem becoming a bit more plane-Couette like. A brief § 7 presents evidence that the centre-mode instability was actually found first in viscoelastic Kolmogorov flow (Boffetta *et al.* 2005) before a final discussion follows in § 8.

2. Formulation

We consider pressure-driven, incompressible channel flow between two walls $y = \pm h$ in the x -direction. Using the half-channel height, h , and the base centreline speed U_{max} to non-dimensionalise the problem, the governing equations become

$$Re \left(\frac{\partial \mathbf{u}}{\partial t} + \mathbf{u} \cdot \nabla \mathbf{u} \right) = -\nabla \mathcal{P} + \beta \nabla^2 \mathbf{u} + \nabla \cdot \mathcal{T}, \quad (2.1)$$

$$\nabla \cdot \mathbf{u} = 0, \quad (2.2)$$

$$\frac{\partial \mathcal{T}}{\partial t} + \mathbf{u} \cdot \nabla \mathcal{T} - 2 \text{sym}(\mathcal{T} \cdot \nabla \mathbf{u}) = -\frac{1}{W} \mathcal{T} + \frac{1 - \beta}{W} (\nabla \mathbf{u} + \nabla \mathbf{u}^T), \quad (2.3)$$

where \mathbf{u} is the velocity field, \mathcal{P} the pressure and \mathcal{T} the polymer stress following Khalid *et al.* (2021a). Here, an Oldroyd-B fluid has been assumed so

$$\mathcal{T} = \frac{1 - \beta}{W} (\mathbf{C} - I), \quad (2.4)$$

where \mathbf{C} is the conformation tensor and I the identity 2nd rank tensor. The parameters of the problem are the Reynolds number, Weissenberg number and the solvent-to-total viscosity ratio

$$Re := \frac{U_{max} h}{\nu}, \quad W := \frac{\lambda U_{max}}{h} \quad \text{and} \quad \beta := \frac{\nu_s}{\nu}, \quad (2.5a-c)$$

respectively, where λ is the microstructural relaxation time, ν_s is the solvent kinematic viscosity and ν is the total kinematic viscosity (following Khalid *et al.* 2021a,b). The scaling of the pressure has been done in anticipation of setting $Re = 0$ in § 4.

The one-dimensional base state is

$$U = U(y)\hat{x} := (1 - y^2)\hat{x}, \quad \nabla P = -2\hat{x}, \quad (2.6)$$

$$\mathbf{T} = \begin{bmatrix} T_{11} & T_{12} \\ T_{12} & T_{22} \end{bmatrix} = (1 - \beta) \begin{bmatrix} 2WU'^2 & U' \\ U' & 0 \end{bmatrix}, \quad (2.7)$$

where $U' := dU/dy$ and, henceforth, the analysis is entirely two-dimensional. The linearised equations for small perturbations

$$\mathbf{v} = u\hat{x} + v\hat{y} := \mathbf{u} - U, \quad p := \mathcal{P} - P \quad \text{and} \quad \boldsymbol{\tau} = \begin{bmatrix} \tau_{11} & \tau_{12} \\ \tau_{12} & \tau_{22} \end{bmatrix} := \boldsymbol{\mathcal{T}} - \mathbf{T}, \quad (2.8a-c)$$

which are all assumed proportional to $e^{ik(x-ct)}$, where $k \in \mathbb{R}$ is a real wavenumber but the frequency $c = c_r + ic_i \in \mathbb{C}$ can be complex ($c_i > 0$ indicates instability; $c_r, c_i \in \mathbb{R}$), are the momentum and incompressibility equations

$$Re [ik(U - c)u + U'v] = -ikp + \beta(D^2 - k^2)u + ik\tau_{11} + D\tau_{12}, \quad (2.9)$$

$$Re [ik(U - c)v] = -Dp + \beta(D^2 - k^2)v + ik\tau_{12} + D\tau_{22}, \quad (2.10)$$

$$iku + Dv = 0, \quad (2.11)$$

for the velocity field and

$$\left[\frac{1}{W} + ik(U - c) \right] \tau_{11} = -vDT_{11} + 2ikT_{11}u + 2T_{12}Du + 2U'\tau_{12} + \frac{2ik(1 - \beta)}{W}u, \quad (2.12)$$

$$\left[\frac{1}{W} + ik(U - c) \right] \tau_{12} = -vDT_{12} + ikT_{11}v + U'\tau_{22} + \frac{1 - \beta}{W}(Du + ikv), \quad (2.13)$$

$$\left[\frac{1}{W} + ik(U - c) \right] \tau_{22} = 2ikT_{12}v + \frac{2(1 - \beta)}{W}Dv, \quad (2.14)$$

for the polymer field, where $D := d/dy$, $T_{11} = 2\Lambda U'^2$, $T_{12} = \Lambda U'/W$, $T_{22} = 0$ and $\Lambda := W(1 - \beta)$ (see (2.7)) in preparation for § 4. The pressure p can be eliminated between (2.9) and (2.10) to produce the vorticity equation

$$\begin{aligned} \beta(D^2 - k^2)^2 v &= -k^2 D(\tau_{11} - \tau_{22}) + ik(D^2 + k^2)\tau_{12} \\ &+ ikRe \left[(U - c)(D^2 - k^2)v - U''v \right]. \end{aligned} \quad (2.15)$$

This equation is good for (asymptotic) analysis but not for a numerical solution where discretising two second-order equations rather than one fourth-order equation is a far better conditioned process.

3. The $Re \rightarrow \infty$ asymptotics for channel flow

A natural starting point for examining the centre-mode instability is to consider the neutral curve in the Re – W plane for fixed β (e.g. figure 2 of Page *et al.* (2020), figure 1 here for $\beta \in \{0.9, 0.98, 0.994\}$ and figure 1 in Dong & Zhang (2022) for pipe flow). The upper and lower branches of this neutral curve have $|Re| \rightarrow \infty$ limits which are now explored.

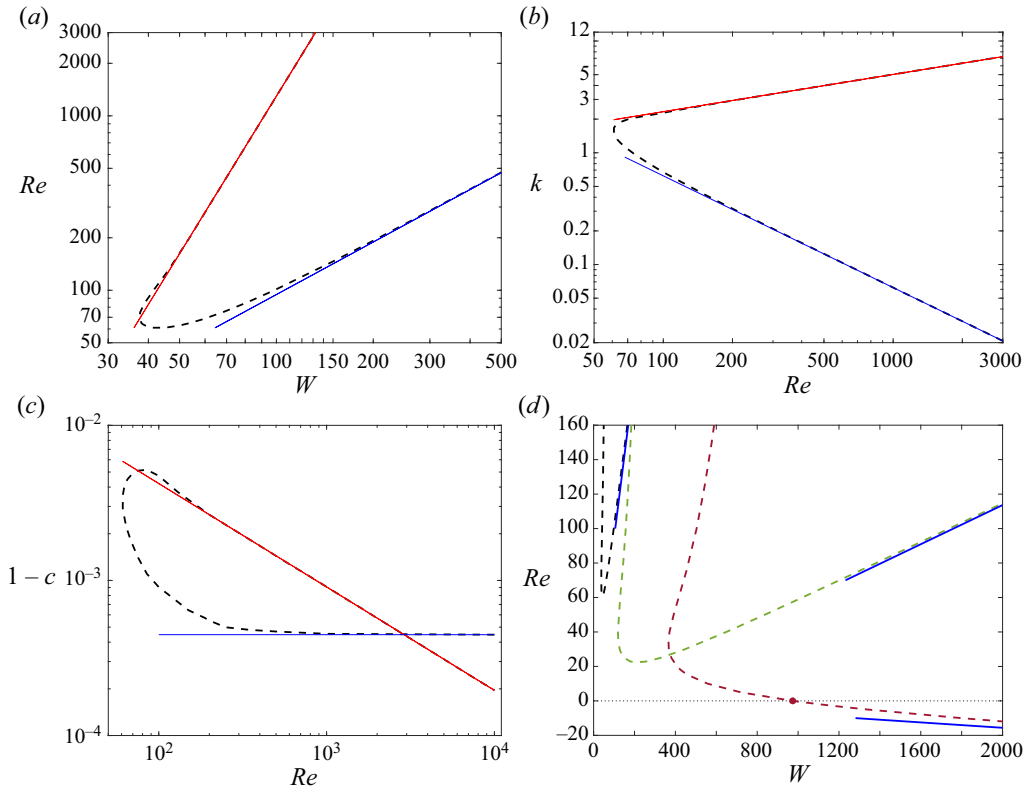


Figure 1. The centre-mode neutral curve (black dashed line) for $\beta = 0.9$ and the asymptotic predictions (upper/lower branch – red/blue solid lines) for W (a), k (b) and $1 - c$ (c). Panel (d) shows how the lower branch of the neutral curve rotates around and crosses the inertialess limit of $Re = 0$ as β increases from $\beta = 0.9$ (black again) through $\beta = 0.98$ (green) to $\beta = 0.994$ (dark red). The solid blue lines indicate the asymptotic prediction for the various lower branches (the $\beta = 0.994$ prediction is reached for $Re \rightarrow -\infty$; note only $Re = O(10)$ is shown here). The dark red dot at $(W, Re) = (973.8, 0)$ is the lowest point reached by the neutral curve for $Re = 0$ for an Oldroyd-B fluid (Khalid *et al.* 2021b). The W – Re neutral curve in (a) is the channel flow equivalent of the pipe flow curve shown on the left in figure 1 of Dong & Zhang (2022).

3.1. Upper branch in Re vs W plane at fixed β

Numerical calculations by Khalid *et al.* (2021a) on the upper branch neutral curve suggest the scaling behaviour

$$(W, k, y, 1 - c) = \left(\frac{\hat{W}}{\delta}, \frac{\hat{k}}{\delta}, \delta \hat{Y}, \hat{a} \delta^2 \right), \quad (3.1)$$

where all hatted variables are $O(\delta^0)$ and $\delta \rightarrow 0$ as $Re \rightarrow \infty$ (the numerical data indicate $\delta = Re^{-1/3}$ but it is worth temporarily ignoring this to reveal a scaling property of the equations). This is the channel flow equivalent of the short-wavelength scalings for pipe flow studied by Dong & Zhang (2022) in their § 4.1. Rescaling the variables (2.8a–c) as follows:

$$(u, v, p, \tau_{11}, \tau_{12}, \tau_{22}) = \left(\hat{u}, \hat{v}, \frac{\hat{p}}{\delta}, \frac{\hat{\tau}_{11}}{\delta}, \frac{\hat{\tau}_{12}}{\delta}, \frac{\hat{\tau}_{22}}{\delta} \right) \quad (3.2)$$

converts (2.9)–(2.14) into

$$Re\delta^3 \left[i\hat{k}(\hat{a} - \hat{Y}^2)\hat{u} - 2\hat{Y}\hat{v} \right] = -i\hat{k}\hat{p} + \beta(\hat{D}^2 - \hat{k}^2)\hat{u} + i\hat{k}\hat{\tau}_{11} + \hat{D}\hat{\tau}_{12}, \quad (3.3)$$

$$Re\delta^3 \left[i\hat{k}(\hat{a} - \hat{Y}^2)\hat{v} \right] = -\hat{D}\hat{p} + \beta(\hat{D}^2 - \hat{k}^2)\hat{v} + i\hat{k}\hat{\tau}_{12} + \hat{D}\hat{\tau}_{22}, \quad (3.4)$$

$$i\hat{k}\hat{u} + \hat{D}\hat{v} = 0, \quad (3.5)$$

$$\begin{aligned} \left[\frac{1}{\hat{W}} + i\hat{k}(\hat{a} - \hat{Y}^2) \right] \hat{\tau}_{11} = & -16(1 - \beta)\hat{W}\hat{Y}\hat{v} + 16i\hat{k}(1 - \beta)\hat{W}\hat{Y}^2\hat{u} \\ & - 4(1 - \beta)\hat{Y}\hat{D}\hat{u} - 4\hat{Y}\hat{\tau}_{12} + \frac{2i\hat{k}(1 - \beta)}{\hat{W}}\hat{u}, \end{aligned} \quad (3.6)$$

$$\begin{aligned} \left[\frac{1}{\hat{W}} + i\hat{k}(\hat{a} - \hat{Y}^2) \right] \hat{\tau}_{12} = & 2(1 - \beta)\hat{v} + 8i\hat{k}(1 - \beta)\hat{W}\hat{Y}^2\hat{v} - 2\hat{Y}\hat{\tau}_{22} \\ & + \frac{1 - \beta}{\hat{W}}(\hat{D}\hat{u} + i\hat{k}\hat{v}), \end{aligned} \quad (3.7)$$

$$\left[\frac{1}{\hat{W}} + i\hat{k}(\hat{a} - \hat{Y}^2) \right] \hat{\tau}_{22} = -4i\hat{k}(1 - \beta)\hat{Y}\hat{v} + \frac{2(1 - \beta)}{\hat{W}}\hat{D}\hat{v}, \quad (3.8)$$

where $\hat{D} := \partial/\partial\hat{Y} = \delta D$ and no terms have been dropped. The polymer equations are therefore invariant under this scaling regardless of δ but $\delta := Re^{-1/3}$ is forced by the momentum equation if inertia and viscous effects are to be balanced in the usual Newtonian way near a critical layer (where $Re(c) = U(y)$). With this choice, no terms are also dropped in the momentum equation so the scaling transformation is exact here for a parabolic base profile. The one change going from the original eigenvalue problem to this scaled version is the position of the boundary which is transformed to $\hat{Y} = \pm\infty$. Solving the asymptotic eigenvalue problem on the neutral curve is then one of finding a neutral eigenfunction which decays away at infinity. In their pipe flow analysis, Dong & Zhang (2022) showed that the decay outside of their central layer is in fact exponential (see their equation (4.8)) so what they analyse as a three-layer structure is actually just one. Another way of seeing this is that the full system (3.3)–(3.8) has only $O(1)$ coefficients.

Given the symmetry of the centre mode (Khalid *et al.* 2021a)

$$(u, v, p, \tau_{11}, \tau_{12}, \tau_{22})(-y) = (u, -v, p, \tau_{11}, -\tau_{12}, \tau_{22})(y), \quad (3.9)$$

it is sufficient to just solve across the lower half-channel, imposing the appropriate symmetry across $y = 0$ and $\hat{u} = \hat{v} = 0$ at some large distance $\hat{Y} = -L$ ($L \gg 1$ with $L = 15$ to 50 used to explore convergence at $\beta = 0.9$); see eigenfunctions in figure 2.

The asymptotic properties $(\hat{W}, \hat{k}, Re(\hat{a}))$ of the upper branch neutral curve are given by seeking

$$\min_{\hat{k}} \{ \hat{W}(\hat{k}) \mid \text{Im}[\hat{a}(\hat{k}, \hat{W})] = 0 \}, \quad (3.10)$$

in the eigenvalue problem (3.3)–(3.8), that is, by finding the smallest value of \hat{W} for which there are no unstable eigenfunctions (the growth rate $kc_i = -\text{Im}(\hat{a})/Re$). The required \hat{k} and \hat{a} are defined by the neutral eigenfunction at this maximum. The results of this

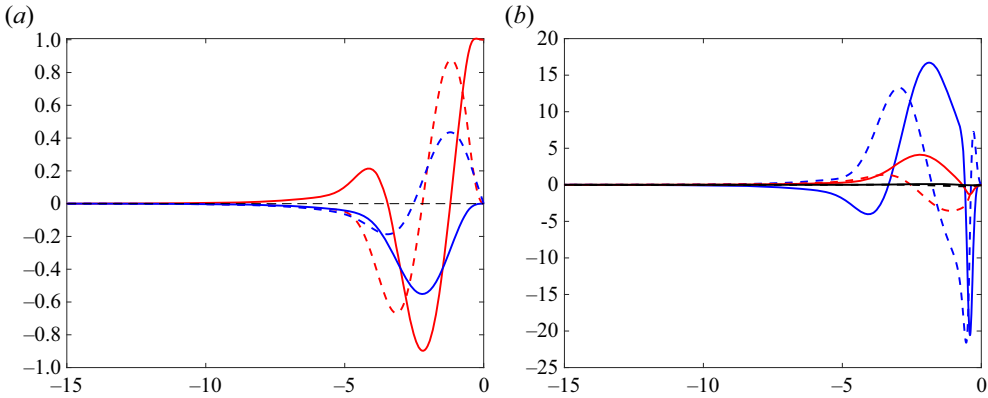


Figure 2. The (asymptotic) eigenfunction on the upper branch neutral curve for $\beta = 0.9$: \hat{u} (red) and \hat{v} (blue) in (a); $\hat{\tau}_{11}$ (blue), $\hat{\tau}_{12}$ (red) and $\hat{\tau}_{22}$ (black) in (b) (in both real/imaginary parts are solid/dashed). The eigenfunction has been normalised so that $\hat{u}(0) = 1$. The calculation has been done over the domain $\hat{Y} \in [-15, 0]$ for clarity but larger domains (e.g. $[-50, 0]$) were used for convergence purposes.

Re	\hat{W}	\hat{k}	\hat{a}
100	8.9409	0.4912	0.1018
200	9.1809	0.5010	0.0908
1000	9.1730	0.5020	0.0909
10 000	9.1727	0.5022	0.0908
∞	9.1725	0.5023	0.0908

Table 1. Upper branch neutral curve characteristics for $\beta = 0.9$ as $Re \rightarrow \infty$. At a given finite Re , $\hat{W} = W/Re^{1/3}$, $\hat{k} = k/Re^{1/3}$ and $\hat{a} = (1 - c)Re^{2/3}$. The bottom line shows the values given in (3.11a–c) found by solving the asymptotic problem in (3.10).

procedure for $\beta = 0.9$ are that

$$W \sim 9.1725Re^{1/3}, \quad k \sim 0.5023Re^{1/3}, \quad c_r \sim 1 - 0.0908Re^{-2/3}, \quad (3.11a-c)$$

on the upper branch neutral curve as $Re \rightarrow \infty$. Table 1 and figure 1 show that this asymptotic result is useful (the curves overlap) down to at least $Re = 150$. Using the elasticity number $E := W/Re$ as in Khalid *et al.* (2021a), these scalings are equivalent to $Re \sim O(E^{-3/2})$, $k \sim O(E^{-1/2})$ and $1 - c \sim O(E)$ as $E \rightarrow 0$ which is consistent with figure 11 in Khalid *et al.* (2021a).

3.2. Lower branch in Re vs W plane at fixed β

Numerical calculations on the lower branch neutral curve (Khalid *et al.* 2021a) suggest a long-wavelength limit scaling of the following form:

$$(u, v, p, \tau_{11}, \tau_{12}, \tau_{22}, W, k) = \left(\hat{u}, \frac{\hat{v}}{Re}, Re\hat{p}_0 + \hat{p}_1 + O(Re^{-1}), Re\hat{\tau}_{11}, \hat{\tau}_{12}, \frac{\hat{\tau}_{22}}{Re}, Re\hat{W}, \frac{\hat{k}}{Re} \right), \quad (3.12)$$

where all hatted variables are $O(1)$ as $Re \rightarrow \infty$, \hat{p}_0 is a constant ($D\hat{p}_0 = 0$) and c stays $O(1)$ and bounded away from 0 (the wall advection speed) and 1 (the centreline advection

β	\hat{W}	\hat{k}	c
0.9	1.058	62.17	0.999553
0.98	17.60	13.32	0.998788
0.994	-128.1	-5.97	0.999198

Table 2. Lower branch neutral curve characteristics for $\beta = 0.9$ and 0.98 as $Re \rightarrow \infty$ and for $\beta = 0.994$ as $Re \rightarrow -\infty$ (hence \hat{W} and \hat{k} are both negative). A comparison with results from finite Re calculations is shown in figure 1(d).

speed); see figure 1(c). In their long-wavelength analysis for pipe flow (their § 3), Dong & Zhang (2022) only consider $1/Re \ll k \ll 1$ and so do not treat the lower branch neutral curve where again $k = O(1/Re)$ is found numerically (Garg *et al.* 2018). With these rescalings, (2.9)–(2.14) become, to leading order,

$$i\hat{k}(1 - y^2 - c)\hat{u} - 2y\hat{v} = -i\hat{k}\hat{p}_0 + \beta D^2\hat{u} + i\hat{k}\hat{\tau}_{11} + D\hat{\tau}_{12} \quad (3.13)$$

$$i\hat{k}(1 - y^2 - c)\hat{v} = -D\hat{p}_1 + \beta D^2\hat{v} + i\hat{k}\hat{\tau}_{12} + D\hat{\tau}_{22}, \quad (3.14)$$

$$i\hat{k}\hat{u} + D\hat{v} = 0, \quad (3.15)$$

for the velocity field and, for the polymer stress,

$$\left[\frac{1}{\hat{W}} + i\hat{k}(1 - y^2 - c) \right] \hat{\tau}_{11} = -16(1 - \beta)\hat{W}_y \hat{v} + 16i\hat{k}(1 - \beta)\hat{W}y^2 \hat{u} - 4y \hat{\tau}_{12} - 4(1 - \beta)yD\hat{u}, \quad (3.16)$$

$$\left[\frac{1}{\hat{W}} + i\hat{k}(1 - y^2 - c) \right] \hat{\tau}_{12} = 2(1 - \beta)\hat{v} + 8i\hat{k}(1 - \beta)\hat{W}y^2 \hat{v} - 2y \hat{\tau}_{22} + \frac{1 - \beta}{\hat{W}}D\hat{u}, \quad (3.17)$$

$$\left[\frac{1}{\hat{W}} + i\hat{k}(1 - y^2 - c) \right] \hat{\tau}_{22} = -4i\hat{k}(1 - \beta)y \hat{v} + \frac{2(1 - \beta)}{\hat{W}}D\hat{v}. \quad (3.18)$$

Since $D\hat{p}_0 = 0$, differentiating (3.13) leads directly to the vorticity equation

$$\beta D^4\hat{v} = -\hat{k}^2 D\hat{\tau}_{11} + i\hat{k}D^2\hat{\tau}_{12} + i\hat{k} \left[(1 - y^2 - c)D^2\hat{v} + 2\hat{v} \right], \quad (3.19)$$

and (3.14), which just defines \hat{p}_1 , can be ignored. The problem defined by (3.15)–(3.19) is then an eigenvalue problem for c . Since there is no rescaling of the spatial dimension, the neutral eigenfunction is global and easily resolved. The asymptotic properties (\hat{W} , \hat{k} , c) of the lower branch neutral curve are given by seeking

$$\max_{\hat{k}} \{ \hat{W}(\hat{k}) \mid \text{Im}[c(\hat{k}, \hat{W})] = 0 \}, \quad (3.20)$$

and the results are shown in table 2. The asymptotic scalings for $\beta = 0.9$ where $Re \rightarrow \infty$ are the same as for $\beta = 0.994$ where $Re \rightarrow -\infty$ but the eigenfunctions look distinctly different in the polymer stress field – see figure 3.

The lower branch scalings are apparent in figure 13 of Khalid *et al.* (2021a) (see also their figure 18). The upper branch asymptote is reached for $Re_c \rightarrow \infty$ and $E \rightarrow 0$ whereas the vertical asymptote $E \rightarrow E_\infty$ (a finite value) as $Re_c \rightarrow \infty$ corresponds to the lower

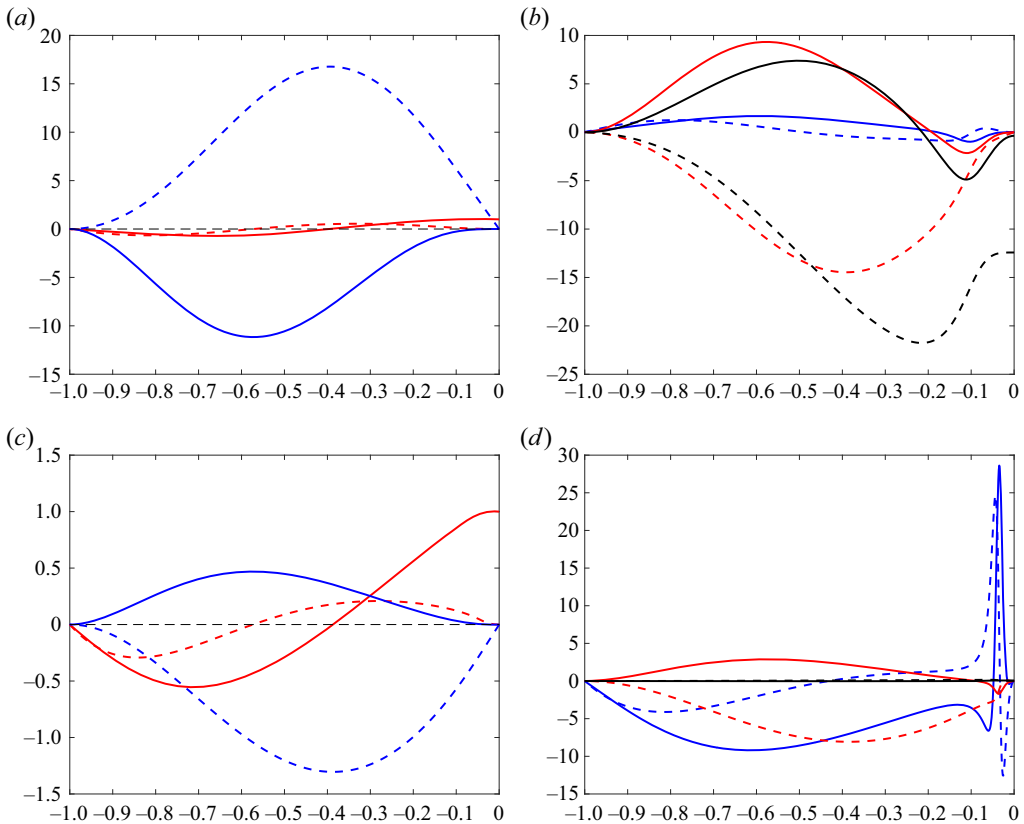


Figure 3. The (asymptotic) eigenfunctions on the lower branch neutral curve for $\beta = 0.9$ (a,b) and $\beta = 0.994$ (c,d): \hat{u} (red) and \hat{v} (blue) in (a,c); $\hat{\tau}_{11}$ (blue), $\hat{\tau}_{12}$ (red) and $\hat{\tau}_{22}$ (black) in (b,d) (in both real/imaginary parts are solid/dashed). Both eigenfunctions have been normalised so that $\hat{u}(0) = 1$.

branch asymptote and the results in table 2 can be used to predict E_c . For example $(1 - \beta)E_c = (1 - \beta)\hat{W} = 0.1058$ at $\beta = 0.9$ and $(1 - \beta)E_c = (1 - \beta)\hat{W} = 0.352$ at $\beta = 0.98$ (note $E_c \lesssim \max_{Re_c} E$ for a given β in figure 13 in Khalid *et al.* 2021a). Khalid *et al.* (2021b) (their figure 4) show that there is no asymptote for $\beta > 0.990552$. Instead the asymptote has to flip to $Re \rightarrow -\infty$ and $E_c < 0$ as shown for example with $\beta = 0.994$ in figure 1(d).

4. The $W \rightarrow \infty$ asymptotics for inertialess ($Re = 0$) channel flow

Without inertia ($Re = 0$), the relevant asymptotic limit is $W \rightarrow \infty$ and $\beta \rightarrow 1$ such that $W(1 - \beta) = \Lambda$ is an $O(1)$ constant, e.g. see insets A and B of figure 2 in Khalid *et al.* (2021b) (or figure 8 in their supplementary material) which suggest $3.5 \lesssim \Lambda \lesssim 10$ for instability. Physically, of course, this means W is large but finite whereas Re can be considered separately as small as desired but is strictly not zero as there is flow. The latter is mathematical: the $Re \rightarrow 0$ limit is regular so it is convenient to set $Re = 0$ to get the true limiting values of key dependencies (e.g. how $1 - c$ scales with W on the neutral curve).

As already mentioned in §3 for $Re > 0$, the centre-mode instability has a certain symmetry about the midplane $y = 0$: u is symmetric and v antisymmetric; see (3.9). Henceforth, we only consider $y \in [-1, 0]$ and impose no-slip boundary conditions

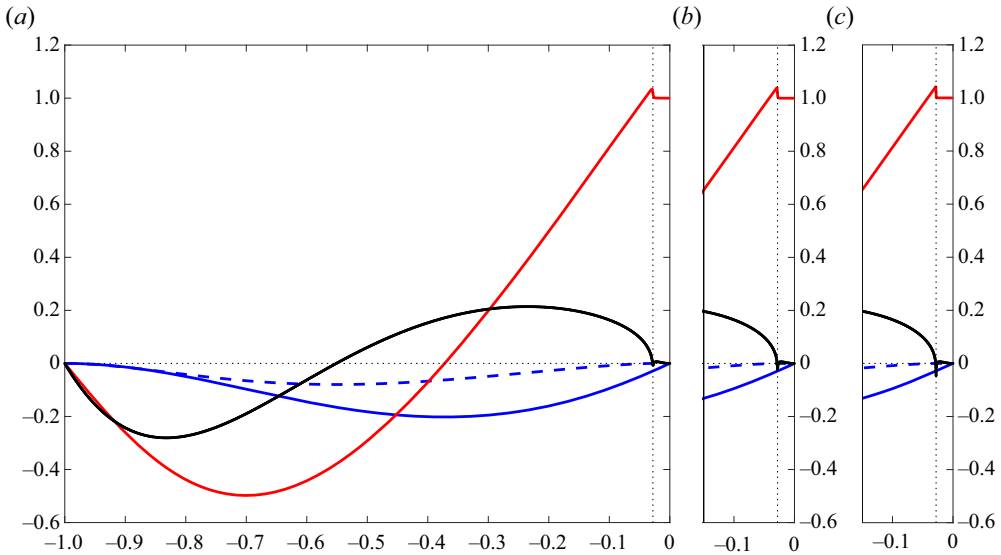


Figure 4. Neutral eigenfunction at $W = 32\,000$ (a), $128\,000$ (b) and $512\,000$ (c), $k = 1.1$ and $\Lambda \approx 4.11 = W(1 - \beta)$ and $c_r \approx 0.99218$. Vertical dotted line near $y = 0$ is the critical layer where $U = 1 - y^2 = c_r$. Here, v is blue (real/imaginary parts solid/dashed, respectively), real part of Dv is red and imaginary part is black. Notice the $O(1)$ jump in $\text{Re}(Dv)$ (and in $\text{Re}(D^2v)$) across the critical layer and the increasingly singular behaviour of $\text{Im}(Dv)$ (black solid line) as W increases (the line dips deeper down at the critical layer approximating the logarithmic singularity).

$v(-1) = Dv(-1) = 0$ at the solid lower plate and symmetry conditions $v(0) = D^2v(0) = 0$ at the midplane. Numerically (see [Appendix A](#) for details), we find on the neutral curve that the eigenfunction has a critical layer near the midplane across which v is continuous but there are jumps in $\text{Re}(Dv)$ and $\text{Re}(D^2v)$ and singular-looking behaviour for $\text{Im}(Dv)$ where the phase of the eigenfunction is set by making $\text{Re}(Dv) = 1$ at the midplane; see [figure 4](#).

A key issue is whether the critical layer at $y = y^*$, defined by $U(y^*) = c_r$ so $y^* := -\sqrt{1 - c_r} \in [-1, 0]$, approaches the midplane, as it does in classical Orr–Sommerfeld analysis for Newtonian shear flows (Drazin & Reid 1981), or not. Certainly, [figure 4](#) suggests ‘not’, and earlier (pre-shooting code) attempts to develop the asymptotic structure could not reconcile $c_r \rightarrow 1$ as $W \rightarrow \infty$ with an $O(1)$ jump in D^2v across the critical layer. This means a novel aspect of the asymptotics here is that, despite $1 - c_r$ being very small, it does in fact remain $O(1)$ as $W \rightarrow \infty$: see [table 3](#).

We introduce a small parameter

$$\varepsilon := \frac{1}{W}, \quad (4.1)$$

and take the distinguished limit $\beta = 1 - \varepsilon\Lambda$, where Λ is an $O(1)$ number to be determined. The eigenfunction plotted in [figure 4](#) shows abrupt changes in the solution as it crosses a critical layer at $y = y^*$. The solution either side of the critical layer – the ‘outer’ solution – must satisfy the governing equations with $\beta = 1$, $\varepsilon = 0$ and $\Lambda = O(1)$ with the critical layer supplying appropriate ‘matching’ conditions between the two parts. The purpose of the asymptotic analysis developed below is to identify analytic expressions for these matching conditions so that the two parts of the outer solution can be fitted

	W	c_r	$k\Lambda_-$	c_r	$k\Lambda_+$
$k = 0.1$	32 000	0.998390	3.41331	0.999893	7.62705
	128 000	0.998408	3.39349	0.999943	9.25320
	512 000	0.998412	3.38849	0.999955	9.76344
	^a 2 048 000	0.998413	3.38724	0.999957	9.80010
	^a 8 192 000	0.998413	3.38693	0.999957	9.81447
	^a 32 768 000	0.998413	3.38686	0.999957	9.81826
	∞	0.998413	3.38683	0.999957	9.81954
$k = 1.1$	64 000	0.9992186	4.526249	0.999853	7.63580
	128 000	0.9992183	4.523676	0.999855	7.65132
	256 000	0.9992181	4.522393	0.999855	7.65932
	^a 1 024 000	0.9992180	4.521432	0.999856	7.66538
	^a 8 192 000	0.9992180	4.521154	0.999856	7.66715
	∞	0.9992180	4.521114	0.999856	7.66740

Table 3. Values of c_r and $k\Lambda := k(1 - \beta)W$ as plotted in inset A of figure 2 in Khalid *et al.* (2021*b*) on the neutral curve for $k = 0.1$ and 1.1 at various large W (\pm indicates upper and lower parts of the neutral curve and ^a results computed using quadruple rather than double precision). The ∞ entry comes from Richardson extrapolation eliminating the leading $O(1/W)$ error. In all cases, Λ is slower to converge than c_r , with the upper curve calculations on the right highlighting this.

together in the limit of $W \rightarrow \infty$ without solving for the critical layer. This defines the leading solution to the problem which includes the leading $O(W^0)$ value of c .

4.1. Outer solution

We refer to the ‘outer’ solution as the solution in the regions $y - y^* = O(1)$. Assuming $v = O(1)$, then τ_{11} and τ_{12} are $O(1)$ whereas $\tau_{22} = O(1/W)$. The leading-order outer problem is then

$$(D^2 - k^2)^2 v = -k^2 D\tau_{11} + ik(D^2 + k^2)\tau_{12}, \quad (4.2)$$

$$iku + Dv = 0, \quad (4.3)$$

$$ik(U - c)\tau_{11} = -vDT_{11} + 2ikT_{11}u + 2U'\tau_{12}, \quad (4.4)$$

$$ik(U - c)\tau_{12} = ikT_{11}v, \quad (4.5)$$

which can be simplified to the fourth-order problem

$$(D^2 - k^2)^2 v = -ikD \left[T_{11}D \left(\frac{v}{U - c} \right) \right] + \frac{ik^3 T_{11}v}{U - c}. \quad (4.6)$$

For $y < y^*$, outer boundary conditions are $v(-1) = 0$, $Dv(-1) = 0$ and, for $y > y^*$, $v(0) = 0 = D^2 v(0)$ with the critical layer supplying 4 matching conditions (for v , Dv , $D^2 v$ and $D^3 v$, respectively). This will produce a well-posed eigenvalue problem for $c(\Lambda, k)$. Ultimately, the problem is to find $\min \Lambda$ (i.e. smallest W at a given β) over all pairs (Λ, k) where $c_i(\Lambda, k) = 0$.

4.2. Inner solution

The inner solution is the solution in the critical layer $y - y^* = O(\varepsilon)$ where the thickness comes from balancing polymer advection and relaxation processes; see the left-hand sides

of the perturbed polymer stress equations (2.12)–(2.14). We therefore define a critical layer variable Y and the corresponding derivative \hat{D} ,

$$Y := \frac{y - y^*}{\varepsilon}, \quad D = \frac{1}{\varepsilon} \hat{D} := \frac{1}{\varepsilon} \frac{\partial}{\partial Y}, \quad (4.7a,b)$$

so that, for example, $1/W + ik(U - c) = \varepsilon(1 + ikU'_*Y + ikU''_*\varepsilon Y^2)$, where $U'_* := U'(y^*) = -2y^* > 0$. Then the appropriate expansions turn out to be

$$v = \hat{v}_0 + \varepsilon \log \varepsilon \Pi Y + \varepsilon \hat{v}_1(Y) + \varepsilon^2 \log \varepsilon \left(\hat{v}_\ell(Y) + \frac{1}{2} \Omega Y^2 \right) + \varepsilon^2 \hat{v}_2(Y) + \varepsilon^3 \log \varepsilon \left(\hat{v}_{2\ell} + \frac{1}{6} \Upsilon Y^3 \right) + \varepsilon^3 \hat{v}_3(Y) + \dots \quad (4.8)$$

$$u = \frac{i}{k} \left[\log \varepsilon \Pi + \hat{D} \hat{v}_1(Y) + \varepsilon \log \varepsilon \left(\hat{D} \hat{v}_\ell(Y) + \Omega Y \right) + \varepsilon \hat{D} \hat{v}_2(Y) + \varepsilon^2 \log \varepsilon \left(\hat{D} \hat{v}_{2\ell} + \frac{1}{2} \Upsilon Y^2 \right) + \varepsilon^2 \hat{D} \hat{v}_3(Y) + \dots \right] \quad (4.9)$$

$$Du = \frac{i}{k} \left[\frac{1}{\varepsilon} \hat{D}^2 \hat{v}_1(Y) + \log \varepsilon \left(\hat{D}^2 \hat{v}_\ell(Y) + \Omega \right) + \hat{D}^2 \hat{v}_2(Y) + \varepsilon \log \varepsilon \left(\hat{D}^2 \hat{v}_{2\ell} + \Upsilon Y \right) + \varepsilon \hat{D}^2 \hat{v}_3(Y) + \dots \right] \quad (4.10)$$

$$D^2 u = \frac{i}{k} \left[\frac{1}{\varepsilon^2} \hat{D}^3 \hat{v}_1(Y) + \frac{\log \varepsilon}{\varepsilon} \hat{D}^3 \hat{v}_\ell(Y) + \frac{1}{\varepsilon} \hat{D}^3 \hat{v}_2(Y) + \log \varepsilon \left(\hat{D}^3 \hat{v}_{2\ell} + \Upsilon \right) + \hat{D}^3 \hat{v}_3(Y) + \dots \right] \quad (4.11)$$

for the velocity fields and

$$\tau_{11} = \frac{1}{\varepsilon^2} \hat{\tau}_{11}^0(Y) + \frac{\log \varepsilon}{\varepsilon} \hat{\tau}_{11}^\ell(Y) + \frac{1}{\varepsilon} \hat{\tau}_{11}^1(Y) + \log \varepsilon \hat{\tau}_{11}^{2\ell} + \hat{\tau}_{11}^2(Y) + \dots, \quad (4.12)$$

$$\tau_{12} = \frac{1}{\varepsilon} \hat{\tau}_{12}^0(Y) + \log \varepsilon \hat{\tau}_{12}^\ell(Y) + \hat{\tau}_{12}^1(Y) + \varepsilon \log \varepsilon \hat{\tau}_{12}^{2\ell} + \varepsilon \hat{\tau}_{12}^2(Y) + \dots, \quad (4.13)$$

$$\tau_{22} = \hat{\tau}_{22}^0(Y) + \varepsilon \log \varepsilon \hat{\tau}_{22}^\ell(Y) + \varepsilon \hat{\tau}_{22}^1(Y) + \varepsilon^2 \log \varepsilon \hat{\tau}_{22}^{2\ell} + \varepsilon^2 \hat{\tau}_{22}^2(Y) + \dots, \quad (4.14)$$

for the polymer stresses, where Π , Ω and Υ are complex constants which will emerge below. The reason we need to go so deep into these expansions is the outer problem is fourth order and therefore requires jump conditions down to $D^3 v$ plus there is a singularity at the critical layer. Together, these require considering the equation for $\hat{D}^3 \hat{v}_3(Y)$ which is $O(\varepsilon^2)$ down in the expansion of $D^3 v$ i.e. we need to go to third order in the expansion. The intermediate $\log \varepsilon$ terms are needed to complement the logarithmic terms which arise in the inner solution otherwise ‘singular’ terms (as $\varepsilon \rightarrow 0$) are forced in the outer solution. These $O(\log \varepsilon)$ terms turn out to be unimportant for deriving the matching conditions.

To keep track of the influence of U' and U'' when we probe later why plane-Couette flow does not have a neutral curve, it is useful to expand the base state around $y = y_*$ in

the critical layer as follows:

$$U = U_* + \varepsilon Y U_*' + \frac{1}{2} \varepsilon^2 Y^2 U_*'' + \dots, \quad (4.15)$$

$$DU = U_*' + \varepsilon Y U_*'' + \dots, \quad (4.16)$$

$$T_{11} = T_{11}^{*(0)} + \varepsilon Y T_{11}^{*(1)} + \varepsilon^2 Y^2 T_{11}^{*(2)} + \dots, \quad (4.17)$$

$$DT_{11} = T_{11}^{*(1)} + 2\varepsilon Y T_{11}^{*(2)} + \dots, \quad (4.18)$$

$$T_{12} = \varepsilon T_{12}^{*(1)} + \varepsilon^2 Y T_{12}^{*(2)} + \dots, \quad (4.19)$$

$$DT_{12} = \varepsilon T_{12}^{*(2)} + \dots, \quad (4.20)$$

where

$$T_{11}^{*(0)} := 2\Lambda U_*'^2, \quad T_{11}^{*(1)} := 4\Lambda U_*' U_*'', \quad T_{11}^{*(2)} := 2\Lambda U_*''^2, \quad (4.21a-c)$$

and

$$T_{12}^{*(1)} := \Lambda U_*', \quad T_{12}^{*(2)} := \Lambda U_*'', \quad (4.22a,b)$$

assuming that $U_*''' = 0$ for simplicity (true for both channel and Couette flow).

Now we substitute expansions (4.8)–(4.11) for the perturbation velocity field, (4.12)–(4.14) for the perturbation polymer stresses and (4.15)–(4.20) for the base state into (2.11)–(2.15) and collect similar-order terms to create a hierarchy of problems in the usual way.

4.3. Leading order: $O(1/\varepsilon^3)$ in the Stokes equation

At leading order

$$X \hat{\tau}_{22}^0 = 2ik T_{12}^{*(1)} \hat{v}_0, \quad (4.23)$$

$$X \hat{\tau}_{12}^0 = ik T_{11}^{*(0)} \hat{v}_0 + U_*' \hat{\tau}_{22}^0, \quad (4.24)$$

$$X \hat{\tau}_{11}^0 = 2U_*' \hat{\tau}_{12}^0, \quad (4.25)$$

$$\hat{D}^4 \hat{v}_1 = -k^2 \hat{D} \hat{\tau}_{11}^0 + ik \hat{D}^2 \hat{\tau}_{12}^0, \quad (4.26)$$

where $X := (1 + ik U_*' Y)$. Solving (4.23)–(4.25) for $\hat{\tau}_{12}^0$ and $\hat{\tau}_{11}^0$ then allows (4.26) to be integrated twice with respect to Y to give

$$D^2 v = \frac{1}{\varepsilon} \hat{D}^2 \hat{v}_1 = -\frac{k^2}{\varepsilon} \int \hat{\tau}_{11}^0 dY + \frac{ik}{\varepsilon} \hat{\tau}_{12}^0 = \frac{k^2 T_{11}^{*(0)} \hat{v}_0}{\varepsilon X} \sim \frac{-ik T_{11}^{*(0)} \hat{v}_0}{U_*' (y - y^*)} \quad \text{as } Y \rightarrow \pm\infty. \quad (4.27)$$

Here, the $O(1/\varepsilon)$ integration constants must be zero otherwise the solution cannot be matched with the outer region where $\varepsilon = 0$ to leading order. This leading inner solution for $D^2 v$ immediately suggests that the asymptotic matching will be a challenge. There is a simple pole singularity in $D^2 v$ at the critical layer and, as a consequence, a double pole singularity in $D^3 v$. Since a double pole is symmetric across the critical layer, it does not enter into the matching conditions for $D^3 v$ but will certainly obscure any matching criterion present involving higher-order less singular behaviour.

Forewarned, we press on and integrate once more to give

$$Dv = \hat{D}\hat{v}_1 = \frac{-ikT_{11}^{*(0)}\hat{v}_0}{U'_*} [\log X + \alpha_1], \quad (4.28)$$

where α_1 is another complex constant. Matching to the exterior requires that a $-ikT_{11}^{*(0)}\hat{v}_0/U'_* \log \varepsilon$ term is present in the inner expression for Dv , otherwise the leading outer solution would depend on $\log \varepsilon$. As a consequence, the first complex coefficient, Π , in the expansions (4.8)–(4.11) has to be $\Pi = -ikT_{11}^{*(0)}\hat{v}_0/U'_* = -2ik\Lambda U'_*\hat{v}_0$ so that

$$Dv \sim \frac{-ikT_{11}^{*(0)}\hat{v}_0}{U'_*} [\log[ikU'_*(y - y^*)] + \alpha_1] \quad \text{for } \varepsilon \ll |y - y^*|, \quad (4.29)$$

and so

$$Dv \rightarrow \begin{cases} \Pi \left(\log |kU'_*(y - y^*)| + \alpha_1 - \frac{1}{2}i\pi \right), & y \rightarrow y^{*-}, \\ \Pi \left(\log |ikU'_*(y - y^*)| + \alpha_1 + \frac{1}{2}i\pi \right), & y \rightarrow y^{*+}, \end{cases} \quad (4.30)$$

is independent of ε . The logarithmic dependence gives a jump in Dv across the critical layer of $i\pi\Pi$. Integrating (4.28) gives

$$\hat{v}_1 = \Pi \left[\frac{X \log X}{ikU'_*} + (\alpha_1 - 1)Y \right], \quad (4.31)$$

since $\hat{v}_1(y_*) = 0$ as $v(y_*) = \hat{v}_0$ by definition.

4.4. Next order: $O(\log \varepsilon/\varepsilon^2)$ in the Stokes equation

Working to next order

$$\hat{\tau}_{22}^\ell = \frac{2ikT_{12}^{*(1)}\Pi Y + 2\Lambda\Pi}{X} = 2\Lambda\Pi, \quad (4.32)$$

$$\hat{\tau}_{12}^\ell = \frac{ikT_{11}^{*(0)}\Pi Y + U'_*\hat{\tau}_{22}^\ell}{X} = 2\Lambda\Pi U'_*, \quad (4.33)$$

$$\hat{\tau}_{11}^\ell = \frac{2\Pi(-T_{11}^{*(0)}X + 2ikU'^2_*T_{12}^{*(1)} + 2U'^2_*\Lambda)}{X^3} = 0, \quad (4.34)$$

so $\hat{D}^4\hat{v}_\ell = -k^2\hat{D}\hat{\tau}_{11}^\ell + ik\hat{D}^2\hat{\tau}_{12}^\ell = 0$ since $\hat{\tau}_{11}^\ell = 0$ and $\hat{\tau}_{12}^\ell$ is a constant. The homogeneous solution for \hat{v}_ℓ is a cubic function of Y but Y^2 or Y^3 dependence would lead to singular outer behaviour as $\varepsilon \rightarrow 0$ (e.g. Y^2 in the u expression (4.9)) and the definition of \hat{v}_0 as v at $Y = 0$ precludes a constant. Hence, \hat{v}_ℓ can only be strictly linear in Y . But this has no bearing on the rest of the calculation as (i) $\hat{D}^3\hat{v}_\ell = 0$ in the equation for $\hat{v}_{2\ell}$ – (4.47) below, and (ii) \hat{v}_ℓ only contributes at $O(\varepsilon \log \varepsilon)$ to v in (4.58) and is neglected to leading order.

4.5. Terms of $O(1/\varepsilon^2)$ in the Stokes equation

This is the order which will give the jump condition in D^2v . At $O(1/\varepsilon^2)$, we get

$$X\hat{\tau}_{22}^1 = -\frac{1}{2}ikU_*''Y^2\hat{\tau}_{22}^0 + 2ikT_{12}^{*(2)}\hat{v}_0 + 2ikT_{12}^{*(1)}\hat{v}_1 + 2\Lambda\hat{D}\hat{v}_1, \quad (4.35)$$

$$X\hat{\tau}_{12}^1 = -\frac{1}{2}ikU_*''Y^2\hat{\tau}_{12}^0 + (ikT_{11}^{*(1)}Y - T_{12}^{*(2)})\hat{v}_0 + ikT_{11}^{*(0)}\hat{v}_1 + U_*'\hat{\tau}_{22}^1 + U_*''Y\hat{\tau}_{22}^0 + \frac{i\Lambda}{k}\hat{D}^2\hat{v}_1, \quad (4.36)$$

$$X\hat{\tau}_{11}^1 = -\frac{1}{2}ikU_*''Y^2\hat{\tau}_{11}^0 - T_{11}^{*(1)}\hat{v}_0 - 2T_{11}^{*(0)}\hat{D}\hat{v}_1 + \frac{2i}{k}T_{12}^{*(1)}\hat{D}^2\hat{v}_1 + 2U_*'\hat{\tau}_{12}^1 + 2U_*''Y\hat{\tau}_{12}^0, \quad (4.37)$$

with

$$\hat{D}^4\hat{v}_2 = \Lambda\hat{D}^4\hat{v}_1 - k^2\hat{D}\hat{\tau}_{11}^1 + ik\hat{D}^2\hat{\tau}_{12}^1. \quad (4.38)$$

Integrating twice

$$\hat{D}^2\hat{v}_2 = \Lambda\hat{D}^2\hat{v}_1 - k^2 \int \hat{\tau}_{11}^1 dY + ik\hat{\tau}_{12}^1 + \Theta Y + \Phi, \quad (4.39)$$

where Θ and Φ are complex constants. Here, ΘY is unmatched in the interior as it would correspond to an $O(1/\varepsilon)$ term in the outer solution so Θ must be 0. In terms of deriving jump conditions across the critical layer, the presence of the constant Φ means we are only interested in the asymptotic behaviour (as $Y \rightarrow \pm\infty$) of the right-hand side of (4.39) which gives rise to jumps across the layer. With this in mind, it is straightforward to show from (4.35) and (4.36) that

$$ik\hat{\tau}_{12}^1 = \frac{k^2[T_{11}^{*(0)}]^2}{U_*'^2}\hat{v}_0 \log X + \text{const.} \quad \text{as } Y \rightarrow \pm\infty. \quad (4.40)$$

The other term on the right-hand side

$$\begin{aligned} -k^2 \int \hat{\tau}_{11}^1 dY = & \int \left\{ \overbrace{\frac{\frac{1}{2}k^2U_*''Y^2\hat{\tau}_{11}^0}{U_*'X}}^{(i)} - \overbrace{\frac{ikT_{11}^{*(0)}\hat{v}_0}{U_*'X}}^{(ii)} - \overbrace{\frac{2ikT_{11}^{*(0)}\hat{D}\hat{v}_1}{U_*'X}}^{(iii)} \right. \\ & \left. - \overbrace{\frac{2T_{12}^{*(1)}\hat{D}^2\hat{v}_1}{U_*'X}}^{(iv)} + \overbrace{\frac{2ikU_*''Y\hat{\tau}_{12}^0}{U_*'X}}^{(v)} + \overbrace{\frac{2ik\hat{\tau}_{12}^1}{X}}^{(vi)} \right\} dX, \end{aligned} \quad (4.41)$$

is more involved, with each labelled term contributing. Respectively, as $Y \rightarrow \pm\infty$,

$$\left. \begin{aligned} (i) & \rightarrow -\frac{ikU_*''T_{11}^{*(0)}}{U_*'^2}\hat{v}_0 \log X, \\ (ii) & \rightarrow -\frac{kT_{11}^{*(1)}}{U_*'}\hat{v}_0 \log X, \\ (iii) & \rightarrow -\frac{2k^2[T_{11}^{*(0)}]^2}{U_*'^2}\hat{v}_0 \left[\frac{1}{2}(\log X)^2 + \alpha_1 \log X \right], \\ (iv) & \rightarrow \frac{2ikU_*''T_{11}^{*(0)}}{U_*'^2}\hat{v}_0 \log X. \end{aligned} \right\} \quad (4.42)$$

The (v) term has to be further subdivided as follows:

$$\begin{aligned} \frac{ik}{U'_*} \int \frac{2U'_* \hat{\tau}_{12}^1}{X} dX = & \int \overbrace{\frac{k^2 U''_* Y^2 \hat{\tau}_{12}^0}{X^2}}^a - \frac{2ik \Lambda U''_* \hat{v}_0}{X^2} + \overbrace{\frac{-2k^2 T_{11}^{*(1)} Y \hat{v}_0}{X^2}}^b \\ & + \underbrace{\frac{-2k^2 T_{11}^{*(0)} \hat{v}_1}{X^2}}_c + \frac{2ik U'_* \hat{\tau}_{22}^1}{X^2} + \frac{2ik U''_* Y \hat{\tau}_{22}^0}{X^2} - \frac{2\Lambda \hat{D}^2 \hat{v}_1}{X^2}, \end{aligned} \quad (4.43)$$

with the respective asymptotic behaviours as $Y \rightarrow \pm\infty$

$$\left. \begin{aligned} (v)_a &\rightarrow -\frac{ik U''_* T_{11}^{*(0)} \hat{v}_0}{U'^2_*} \log X, \\ (v)_b &\rightarrow \frac{2ik T_{11}^{*(1)} \hat{v}_0}{U'_*} \log X, \\ (v)_c &\rightarrow \frac{2k^2 [T_{11}^{*(0)}]^2 \hat{v}_0}{U'^2_*} \left[\frac{1}{2} (\log X)^2 + (\alpha_1 - 1) \log X \right]. \end{aligned} \right\} \quad (4.44)$$

Adding all the contributions together

$$\begin{aligned} \hat{D}^2 \hat{v}_2 &\sim \left[\frac{-k^2 [T_{11}^{*(0)}]^2}{U'^2_*} + \frac{ik T_{11}^{*(1)}}{U'_*} \right] \log X + \text{const.} \\ &= (4ik \Lambda U''_* - 4k^2 \Lambda^2 U'^2_*) \hat{v}_0 \log[ik U'_* (y - y_*)] + \text{const.}, \\ &= \Omega \log[ik U'_* (y - y_*)] + \text{const.} \end{aligned} \quad (4.45)$$

This indicates that the second complex coefficient Ω in the expansions (4.8)–(4.11) has to be $\Omega = (4ik \Lambda U''_* - 4k^2 \Lambda^2 U'^2_*) \hat{v}_0$ to avoid an $O(\log \varepsilon)$ term in the outer leading solution for Du (see (4.10)). More importantly (4.45) also gives the finite jump in $D^2 v$ across the critical layer. Integrating (4.45) twice to complete the solution at this order gives

$$\left. \begin{aligned} \hat{D} \hat{v}_2 &= -\frac{i\Omega}{k U'_*} (X \log X - X) + \alpha_2, \\ \text{and } \hat{v}_2 &= -\frac{\Omega}{k^2 U'^2_*} \left(\frac{1}{2} X^2 \log X - \frac{3}{4} X^2 \right) + \alpha_2 Y + \beta_2, \end{aligned} \right\} \quad (4.46)$$

where α_2 and β_2 are constants. Since $\hat{v}_2 = 0$ at $Y = 0$ ($X = 1$) by the definition of \hat{v}_0 , $\beta_2 = -3\Omega/(4k^2 U'^2_*)$.

4.6. Terms of $O(\log \varepsilon/\varepsilon)$ in the Stokes equation

The $O(\log \varepsilon/\varepsilon)$ Stokes equation balance integrated once gives

$$\hat{D}^3 \hat{v}_{2\ell} = \Lambda \hat{D}^3 \hat{v}_\ell - k^2 \hat{\tau}_{11}^{2\ell} + ik \hat{D} \hat{\tau}_{12}^{2\ell} + \text{const.}, \quad (4.47)$$

with $\hat{D}^3 \hat{v}_\ell = 0$ (see § 4.4). Now, since

$$\left. \begin{aligned} \hat{\tau}_{22}^{2\ell} &= \frac{1}{X} \left(-\frac{1}{2} i k U_*'' Y^2 \hat{\tau}_{22}^\ell + i k \Omega T_{12}^{*(1)} Y^2 + 2 i k T_{12}^{*(2)} Y^2 \Pi + 2 \Lambda \Omega Y \right) \sim O\left(\frac{Y^2}{X}\right), \\ \hat{\tau}_{12}^{2\ell} &= \frac{1}{X} \left(-\frac{1}{2} i k U_*'' Y^2 \hat{\tau}_{12}^\ell + (i k T_{11}^{*(1)} Y - T_{12}^{*(2)}) \Pi Y + \frac{1}{2} i k \Omega T_{11}^{*(0)} Y^2 \right. \\ &\quad \left. + U_*'' Y \hat{\tau}_{22}^\ell + U_*' \hat{\tau}_{22}^{2\ell} + \frac{i \Lambda \Omega}{k} \right) \sim O\left(\frac{Y^2}{X}\right), \\ \hat{\tau}_{11}^{2\ell} &= \frac{1}{X} \left(-3 \Pi Y T_{11}^{*(1)} - 2 T_{11}^{*(0)} \Omega Y + \frac{2 i T_{12}^{*(1)} \Omega}{k} + 2 U_*'' Y \hat{\tau}_{12}^\ell + 2 U_*' \hat{\tau}_{12}^{2\ell} \right) \sim O\left(\frac{Y^2}{X^2}, \frac{Y}{X}\right), \end{aligned} \right\} \quad (4.48)$$

as $|Y| \rightarrow \infty$, the right-hand side of (4.47) generates at best constant terms as $Y \rightarrow \pm\infty$, which can be removed by the ‘const’. Hence, there is no consequence outside the critical layer as expected (the $\log \varepsilon$ terms are there just to fix up the logarithmic terms in the inner solution). Note, however, $\hat{v}_{2\ell}$ is non-trivial in the critical layer but it just does not drive a jump across it.

4.7. Terms of $O(1/\varepsilon)$ in the Stokes equation

This is the order at which a finite jump in $D^3 v$ across the critical layer is determined. The $O(1/\varepsilon)$ Stokes equation balance integrated once gives

$$\hat{D}^3 \hat{v}_3 = \underbrace{2k^2 \hat{D} \hat{v}_1}_{(a)} + \Lambda \hat{D}^3 \hat{v}_2 - \underbrace{k^2 \hat{\tau}_{11}^2}_{(d)} + \underbrace{k^2 \hat{\tau}_{22}^0}_{(c)} + \underbrace{i k \hat{D} \hat{\tau}_{12}^2}_{(b)} + i k^3 \int \hat{\tau}_{12}^0 dY. \quad (4.49)$$

Since we are only interested in jumps across the critical layer, we focus on the logarithmic terms appearing on the right-hand side of (4.49) (only the labelled terms contribute). Terms (a) and (b) follow immediately

$$\left. \begin{aligned} (a) &\sim -\frac{2 i k^3 T_{11}^{*(0)} \hat{v}_0}{U_*'} \log X \quad (\text{from (4.28)}), \\ (b) &\sim \frac{i k^3 T_{11}^{*(0)} \hat{v}_0}{U_*'} \log X \quad (\text{from (4.24)}), \end{aligned} \right\} \quad (4.50)$$

as $|Y| \rightarrow \infty$. Terms (c) and (d) require more work going to yet higher order in the polymer stress equations (2.12)–(2.14). Starting with (c)

$$\begin{aligned} i k \hat{D} \hat{\tau}_{12}^2 &= \hat{D} \left[\frac{\frac{1}{2} k^2 U_*'' Y^2 \hat{\tau}_{12}^{*(1)}}{X} \right] + \hat{D} \left[\frac{(-k^2 T_{11}^{*(1)} Y - i k T_{12}^{*(2)}) \hat{v}_1}{X} \right] + \hat{D} \left[-\frac{k^2 T_{11}^{*(0)} \hat{v}_2}{X} \right] \\ &\quad + \hat{D} \left[\frac{i k U_*' \hat{\tau}_{22}^2}{X} + \frac{-k^2 (\Lambda + Y^2 T_{11}^{*(2)}) \hat{v}_0}{X} + \frac{i k U_*'' Y \hat{\tau}_{22}^1}{X} - \frac{\Lambda \hat{D}^2 \hat{v}_2}{X} \right]. \end{aligned} \quad (4.51)$$

Then the labelled terms have the following logarithmic behaviour:

$$\left. \begin{aligned} \text{(i)} &\sim -\frac{k^2 U_*'' [T_{11}^{*(0)}]^2 \hat{v}_0}{2U_*'^3} \log X, \\ \text{(ii)} &\sim \frac{k^2 T_{11}^{*(0)} T_{11}^{*(1)} \hat{v}_0}{U_*'^2} \log X, \\ \text{(iii)} &\sim \frac{ik T_{11}^{*(0)} \Omega}{2U_*'} \log X, \end{aligned} \right\} \quad (4.52)$$

as $|Y| \rightarrow \infty$. Now, considering (d),

$$\begin{aligned} -k^2 \hat{\tau}_{11}^2 &= \frac{\frac{1}{2} ik^3 U_*'' Y^2 \hat{\tau}_{11}^1}{X} + \frac{2k^2 T_{11}^{*(2)} Y \hat{v}_0}{X} + \underbrace{\frac{k^2 T_{11}^{*(1)} \hat{v}_1}{X}}^{(a)} + \underbrace{\frac{k^2 T_{11}^{*(0)} \hat{D} \hat{v}_2}{X}}^{(b)} + \underbrace{\frac{k^2 T_{11}^{*(1)} Y \hat{D} \hat{v}_1}{X}}^{(c)} \\ &\quad - 2 \frac{ik^2 T_{12}^{*(1)} \hat{D}^2 \hat{v}_2}{X} - 2 \frac{ik^2 T_{12}^{*(2)} Y \hat{D}^2 \hat{v}_1}{X} + \underbrace{\frac{-2k^2 U_*' \hat{\tau}_{12}^2}{X}}_{(d)} + \underbrace{\frac{-2k^2 U_*'' Y \hat{\tau}_{12}^1}{X}}_{(e)}, \end{aligned} \quad (4.53)$$

with labelled terms having the following logarithmic behaviour:

$$\left. \begin{aligned} (a) &\sim -\frac{k^2 T_{11}^{*(0)} T_{11}^{*(1)} \hat{v}_0}{U_*'^2} \log X, \\ (b) &\sim -\frac{2ik T_{11}^{*(0)} \Omega}{U_*'} \log X, \\ (c) &\sim -\frac{2k^2 T_{11}^{*(0)} T_{11}^{*(1)} \hat{v}_0}{U_*'^2} \log X, \\ (d) &\sim \left[-\frac{k^2 U_*'' [T_{11}^{*(0)}]^2 \hat{v}_0}{U_*'^3} + \frac{2k^2 T_{11}^{*(0)} T_{11}^{*(1)} \hat{v}_0}{U_*'^2} + \frac{ik T_{11}^{*(0)} \Omega}{U_*'} \right] \log X, \\ (e) &\sim \frac{2k^2 U_*'' [T_{11}^{*(0)}]^2 \hat{v}_0}{U_*'^3} \log X, \end{aligned} \right\} \quad (4.54)$$

as $|Y| \rightarrow \infty$. Bringing the expressions from (4.50), (4.52) and (4.54) together

$$\hat{D}^3 \hat{v}_3 \sim \Upsilon \log X := \left[k\Pi(k - 2i\Lambda) - ik\Lambda U_*' \Omega \right] \log X \quad \text{as } Y \rightarrow \pm\infty, \quad (4.55)$$

which gives a finite jump in $D^3 v$ across the critical layer. To complete the solution at this order, integrating repeatedly

$$\left. \begin{aligned} \hat{D}^2 \hat{v}_3 &= -\frac{i\Upsilon}{kU_*'} (X \log X - X) + \alpha_3, \\ \hat{D} \hat{v}_3 &= -\frac{\Upsilon}{k^2 U_*'^2} \left(\frac{1}{2} X^2 \log X - \frac{3}{4} X^2 \right) + \alpha_3 Y + \beta_3, \\ \hat{v}_3 &= \frac{i\Upsilon}{k^3 U_*'^3} \left(\frac{1}{6} X^3 \log X - \frac{11}{36} X^3 \right) + \frac{1}{2} \alpha_3 Y^2 + \beta_2 Y + \gamma_3, \end{aligned} \right\} \quad (4.56)$$

where α_3, β_3 and γ_3 are constants with γ_3 set by ensuring that $\hat{v}_3 = 0$ at $Y = 0$ ($X = 1$).

4.8. Matching conditions across the critical layer

All the work above has built up a high-order (in ε) representation for v as follows:

$$v = \hat{v}_0 + \varepsilon [\hat{v}_1(Y) + \Pi Y \log \varepsilon] + \varepsilon^2 \left[\hat{v}_2(Y) + \left(\hat{v}_\ell(Y) + \frac{1}{2} \Omega Y^2 \right) \log \varepsilon \right] + \varepsilon^3 \log \varepsilon \hat{v}_{2\ell} \\ + \varepsilon^3 \left[\hat{v}_3(Y) + \frac{1}{6} \Upsilon Y^3 \log \varepsilon \right] + \dots \quad (4.57)$$

What is important is the behaviour as $y \rightarrow y^*$ for matching to the outer solutions; specifically

$$v = \hat{v}_0 + \left[\Pi(y - y^*) + \frac{1}{2} \Omega(y - y^*)^2 + \frac{1}{6} \Upsilon(y - y^*)^3 + \dots \right] \log[ikU'_*(y - y^*)] \\ + \Pi(y - y^*)(\alpha_1 - 1) - \frac{3}{4} \Omega(y - y^*)^2 - \frac{11}{36} \Upsilon(y - y^*)^3 + O(\varepsilon \log \varepsilon, (y - y^*)^4), \quad (4.58)$$

where all the constants disappear at leading order in ε with the exception of α_1 . From this, the various jump conditions

$$[A]^\pm := A(y^* + \delta) - A(y^* - \delta), \quad (4.59)$$

where $\delta \rightarrow 0$, can be deduced as

$$[v]^\pm = 2\Pi\delta[\log|kU'_*\delta| + \alpha_1 - 1] + \frac{1}{2}i\pi\Omega\delta^2 + \Upsilon\delta^3 \left[\frac{1}{3} \log|kU'_*\delta| - \frac{11}{18} \right], \quad (4.60)$$

$$[Dv]^\pm = i\pi\Pi + 2\delta\Omega[\log|kU'_*\delta| - 1] + \frac{1}{2}i\pi\Upsilon\delta^2, \quad (4.61)$$

$$[D^2v]^\pm = \frac{2\Pi}{\delta} + i\pi\Omega + 2\delta\Upsilon[\log|kU'_*\delta| - 1], \quad (4.62)$$

$$[D^3v]^\pm = \frac{2\Omega}{\delta} + i\pi\Upsilon, \quad (4.63)$$

with

$$\Pi = -2ik\Lambda U'_* \hat{v}_0, \quad \Omega = (4ik\Lambda U''_* - 4k^2\Lambda^2 U'^2_*) \hat{v}_0, \quad \Upsilon = k\Pi(k - 2i\Lambda) - ik\Lambda U'_* \Omega. \quad (4.64a-c)$$

These matching conditions (4.60)–(4.63) are the culmination of the inner solution analysis. They are correct to 3 orders in δ which is clear in all but the last jump condition. Here, $D^3v \sim -\Pi/\delta^2$ actually which, since it is even in δ , does not appear in the required jump.

The outer problem (4.6) is fourth order with 2 boundary conditions at each boundary (no slip at $y = -1$ and symmetry conditions at $y = 0$). So matching the outer solutions across the critical layer requires 4 (complex) conditions to determine 4 complex constants. Since the problem is linear and so the amplitude and phase are indeterminate, one of these conditions instead sets the two real numbers Λ and c (as $c_i = 0$ on the neutral curve). However, there is an extra (complex) unknown in the jump conditions (4.60)–(4.63), α_1 , which means a fifth (complex) condition is needed.

In practice, it is convenient to choose $\hat{v}_0 = -1$ to set the amplitude and phase of the eigenfunction, thereby upping the matching requirement to 6 complex equations (for the 4 complex constants specifying the two outer solutions plus the complex number α_1 and

real pair (Λ, c)). An extra equation comes from now being able to impose the behaviour of v

$$v^- + \delta \Pi \alpha_1 = -1 + \left(-\delta \Pi + \frac{1}{2} \delta^2 \Omega - \frac{1}{6} \delta^3 \Upsilon\right) \left[\log |k U'_* \delta| - \frac{1}{2} i \pi\right] + \delta \Pi - \frac{3}{4} \delta^2 \Omega + \frac{11}{36} \delta^3 \Upsilon, \quad (4.65)$$

$$v^+ - \delta \Pi \alpha_1 = -1 + \left(\delta \Pi + \frac{1}{2} \delta^2 \Omega + \frac{1}{6} \delta^3 \Upsilon\right) \left[\log |k U'_* \delta| + \frac{1}{2} i \pi\right] - \delta \Pi - \frac{3}{4} \delta^2 \Omega - \frac{11}{36} \delta^3 \Upsilon, \quad (4.66)$$

as the critical layer is approached from either side ($v^\pm = v(y^* \pm \delta)$). The final extra equation comes from also doing the same for Dv

$$Dv^- - \Pi \alpha_1 = \left(\Pi - \delta \Omega + \frac{1}{2} \delta^2 \Upsilon\right) \left[\log |k U'_* \delta| - \frac{1}{2} i \pi\right] + \delta \Omega - \frac{3}{4} \delta^2 \Upsilon, \quad (4.67)$$

$$Dv^+ - \Pi \alpha_1 = \left(\Pi + \delta \Omega + \frac{1}{2} \delta^2 \Upsilon\right) \left[\log |k U'_* \delta| + \frac{1}{2} i \pi\right] - \delta \Omega - \frac{3}{4} \delta^2 \Upsilon, \quad (4.68)$$

which crucially does not introduce any new unknown constants. These 4 conditions together with the jump conditions (4.62) and (4.63) give the required 6 conditions.

In practice, we impose the conditions (4.63)–(4.68) to specify the velocity fields and α_1 , and then integrate the outer solutions from the lower wall at $y = -1$ and centreline at $y = 0$ to the critical layer. The remaining (complex) $D^2 v$ jump condition (4.62) at the critical layer then defines a complex scalar function of Λ and c_r which must vanish. This is arranged by applying Newton Raphson to the variables Λ and c given a good enough initial guess for them. The matching is not surprisingly quite delicate because up to the third derivative in v has to be matched, there is singular behaviour and the critical layer can get very close to the centreline which imposes the condition $\delta \ll \sqrt{1-c}$. Invariably, δ needs to be smaller than 10^{-4} to see convergence which, since terms up to $O(\delta^3)$ need to be resolved, requires quadruple precision arithmetic. Table 4 shows the results of matching at $k = 0.1$, which is away from the neutral curve nose (see inset A of figure 2 of Khalid *et al.* 2021b), and $k = 1.1$, which is close to it. For the upper neutral curve at $k = 0.1$ where $c = 0.99957$, the critical layer is so close to the centreline that only matching with quadruple precision is possible.

Figure 5 plots the error in estimating the (lower neutral curve) limiting values of c and Λ at $k = 1.1$ via the numerical solution taking $W \rightarrow \infty$ and the asymptotic matching approach taking $\delta \rightarrow 0$. The asymptotically matched outer velocity fields over $y \in [-1, y^* - \delta] \cup [y^* + \delta, 0]$ compare excellently with the full numerical solution at $W = 32\,000$ shown in figure 4 – see figure 6. Reducing δ from 2×10^{-4} to 5×10^{-6} shows the singularity in $\text{Im}(Dv)$ at the critical layer when the phase of the eigenfunction is set by $Dv = 1$ at the midplane. The numerically computed stress field at $W = 32\,000$ is compared in figure 7 with the matched outer solution (using $\delta = 2 \times 10^{-4}$) and the leading inner asymptotic solution, again with excellent agreement.

The key realisation from the asymptotic analysis is that the structure of the critical layer is built upon a non-vanishing cross-stream velocity there. This is reflected in the fact that everything scales with it – specifically \hat{v}_0 in the expansion (4.8). For example, the 3 matching constants in (4.64a–c) are proportional to \hat{v}_0 ; if these are zero, the critical layer has no effect on the outer solutions. The cross-stream velocity, however, must vanish at the midplane by the symmetry conditions and so there has to be an $O(1)$ outer layer between the critical layer and the centreline to bring about this adjustment (derivatives in the outer regions are $O(1)$ and $\hat{v}_0 = O(1)$ to set the normalisation of the eigenfunction).

	δ	c_r	$k\Lambda_-$	c_r	$k\Lambda_+$
$k = 0.1$	32×10^{-4}	0.99855267713	3.446546	—	—
	8×10^{-4}	0.99844946362	3.400230	—	—
	2×10^{-4}	0.99842217083	3.389936	—	—
	$^a 1 \times 10^{-4}$	0.99841758893	3.388332	0.99995720674	9.856305
	$^a 1 \times 10^{-5}$	0.99841347771	3.386965	0.99995700930	9.823193
	$^a 1 \times 10^{-6}$	0.99841306978	3.386842	0.99995698930	9.819909
	$^a 3 \times 10^{-7}$	0.99841304032	3.386833	0.99995698769	9.819643
	0	0.99841302769	3.386829	0.99995698700	9.819529
$k = 1.1$	32×10^{-4}	0.99923583992	4.525599	0.99988318362	8.328840
	8×10^{-4}	0.99922208240	4.519946	0.99986369555	7.823761
	2×10^{-4}	0.99921892090	4.520468	0.99985801319	7.706180
	$^a 1 \times 10^{-4}$	0.99921842343	4.520720	0.99985702092	7.686784
	$^a 1 \times 10^{-5}$	0.99921799797	4.521053	0.99985611212	7.669353
	$^a 1 \times 10^{-6}$	0.99921795926	4.521106	0.99985602002	7.667615
	$^a 1 \times 10^{-7}$	0.99921795576	4.521113	0.99985600515	7.667345
	0	0.99921795537	4.521114	0.99985600350	7.667315

Table 4. Values of c_r and $k\Lambda$ found by asymptotic matching on the neutral curve for $k = 0.1$ and 1.1 with various choices of δ (\pm indicates upper and lower parts of the neutral curve and a results computed using quadruple rather than double precision). The quadruple precision calculations were done using the multi-precision extension package ‘Advanpix’ for Matlab. The ‘0’ predictions are found by applying Richardson extrapolation to the last two entries assuming a leading $O(\delta)$ error. All four cases compare well with the numerical predictions marked by ‘ ∞ ’ in [table 3](#).

This explains why the critical layer cannot approach the centreline as $W \rightarrow \infty$ or equivalently why c converges to a finite value which is not 1. It remains unclear why this finite value is numerically so close to 1 (e.g. $1 - c = 4.3 \times 10^{-5}$ for $k = 0.1$ in [table 3](#)) but the plausible hypothesis is that the critical layer can only manifest in a low shear region compared with the rest of the domain. This is probed a little in § 6 below but first we give a discussion on the instability mechanism.

5. Instability mechanism

The asymptotic analysis above separates the ‘inner’ solution in the critical layer from the ‘outer’ solution, allowing scrutinisation of how the instability manifests in the latter. The role of the critical layer is then viewed as generating ‘energising’ internal conditions for the outer solution (or boundary conditions for the two parts of the outer solution). Before proceeding in this manner, we simplify the outer equations by rewriting some terms using the streamline displacement $\phi := v/ik(U - c)$ instead of v (e.g. Rallison & Hinch 1995) to get

$$0 = -ikp + (D^2 - k^2)u - ikT_{11}D\phi, \quad (5.1)$$

$$0 = -Dp + (D^2 - k^2)v - k^2T_{11}\phi, \quad (5.2)$$

$$0 = iku + Dv, \quad (5.3)$$

$$\tau_{11} = -2T_{11}D\phi - \phi DT_{11}, \quad (5.4)$$

$$\tau_{12} = ikT_{11}\phi. \quad (5.5)$$

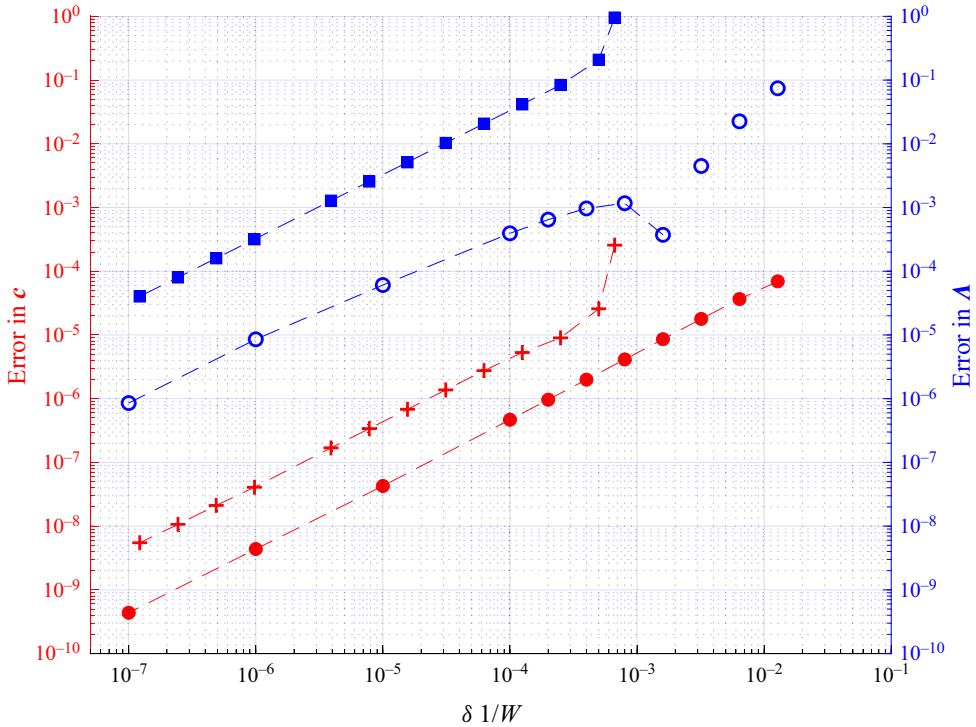


Figure 5. The ‘error’ in c and Λ as a function of finite δ compared with their $\delta = 0$ limiting values for the asymptotically matched solution and for the numerical solution as a function of $1/W$ compared with the $W \rightarrow \infty$ limit for $k = 1.1$ and $(c, \Lambda) = (0.999218, 4.1101)$. Circles are used for the asymptotically matched solutions (filled red circles for c and open blue circles for Λ); for the numerical approximations, red + is used for c and blue filled squares for Λ . The ‘truth’ $(c(0), \Lambda(0))$ is estimated by assuming $c(0) = c(\delta) + a\delta + b\delta^2 + \dots$ and eliminating the leading error between the two most accurate predictions – i.e. $c(0) := (10c(10^{-7}) - c(10^{-6}))/9$. Note $1/W \leq 1/974$ for an Oldroyd-B fluid (Khalid *et al.* 2021*b*).

While the above analysis shows that v is continuous across the critical layer, $\phi \sim 1/(U - c)$ as the critical layer is approached. Hence, the streamline displacement is maximal there and the question is how this drives the polymer field which must in turn offset the viscous dissipation in the inertialess momentum equation.

Approaching the neutral curve, $c_i \rightarrow 0$, means that ϕ is exactly out of phase with v so that $-k^2 T_{11} \phi$ can do no work in energising v in (5.2), that is,

$$\frac{k}{2\pi} \int_0^{2\pi/k} \text{Re}[v e^{ik(x-ct)}] \text{Re}[-k^2 T_{11} \phi e^{ik(x-ct)}] dx = 0. \quad (5.6)$$

This means the viscous dissipation in the cross-stream variable v can only be offset by the pressure term and the polymer stress driving of the velocity field must occur in (5.1). To confirm this, the power input

$$P(y) := \frac{k}{2\pi} \int_0^{2\pi/k} \text{Re}[u e^{ik(x-ct)}] \text{Re}[-ik T_{11} D\phi e^{ik(x-ct)}] dx, \quad (5.7)$$

is plotted in figure 8, which clearly shows that the polymer stress energising of u is localised at the critical layer. The specific solution used here is shown in figure 9.

The polymer stress equations are particularly simple when written in terms of the streamline displacement and are interpretable. The first term on the right of (5.4) represents

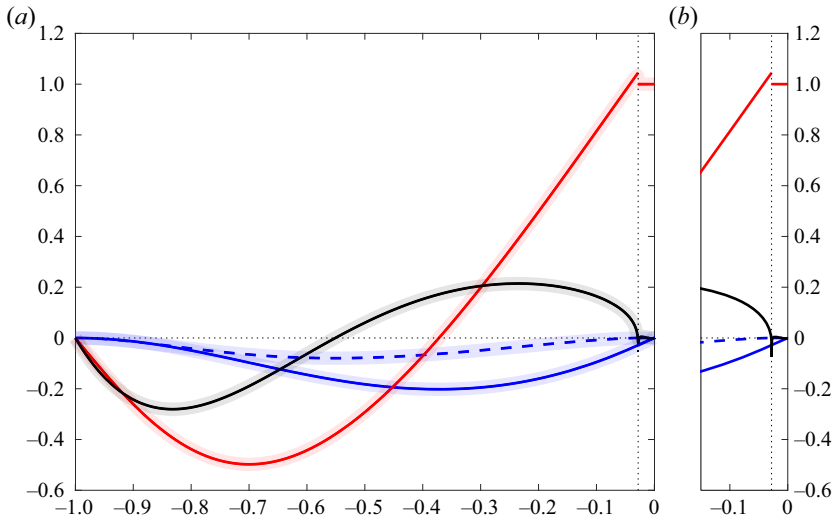


Figure 6. The outer solution at $k = 1.1$ and $\Lambda \approx 4.11 = W(1 - \beta)$ found by applying the matching conditions (4.63)–(4.68) with $\delta = 2 \times 10^{-4}$ (a) and $\delta = 5 \times 10^{-6}$ (b). In (a), the numerical solution shown in figure 4 at $W = 32\,000$ is plotted using thick translucent lines of the appropriate colour – v is blue (real/imaginary parts solid/dashed respectively), real part of Dv is red and imaginary part is black – to demonstrate that the asymptotic and numerical solutions match well. Panel (b) shows the increasingly singular behaviour in the imaginary part of Dv at the critical point as δ decreases.

the increase in the streamwise-normal stress due to streamline compression (in y), the second term represents the maintenance of the initial basic stress on displaced streamlines and the third term (on the right-hand side of (5.5)) is simply the generation of tangential polymer stress due to tilting of the base polymer stress lines (Rallison & Hinch 1995). Multiplying (5.4) by $ik(U - c)$ recovers the time-derivative and advection terms on the left-hand side of (5.4) and thereby recovers the proper driving terms on the right-hand side

$$ik(U - c)\tau_{11} = -2ik(U - c)T_{11}D\phi - ik(U - c)\phi DT_{11}. \quad (5.8)$$

Their energising effects

$$C(y) := \frac{k}{2\pi} \int_0^{2\pi/k} \text{Re}[\tau_{11}e^{ik(x-ct)}] \text{Re}[-2ik(U - c)T_{11}D\phi e^{ik(x-ct)}] dx, \quad (5.9)$$

and

$$D(y) := \frac{k}{2\pi} \int_0^{2\pi/k} \text{Re}[\tau_{11}e^{ik(x-ct)}] \text{Re}[-ik(U - c)\phi DT_{11}e^{ik(x-ct)}] dx, \quad (5.10)$$

are shown in figure 8. Both terms are global and barely register the critical layer with streamline compression causing the streamwise-normal stress to increase ($C(y)$) while displacement across the base shear field ($D(y)$) works negatively to balance it on the neutral curve. It is worth remarking that the reintroduction of the factor $(U - c)$ into (5.8) is responsible for desensitising $C(y)$ to the critical layer as otherwise a significant component $-T_{11}D\phi$ – is the same as that in P .

The mechanism of the instability can therefore be seen as one in which the critical layer acts like a pair of ‘bellows’ periodically sucking the flow streamlines together – see

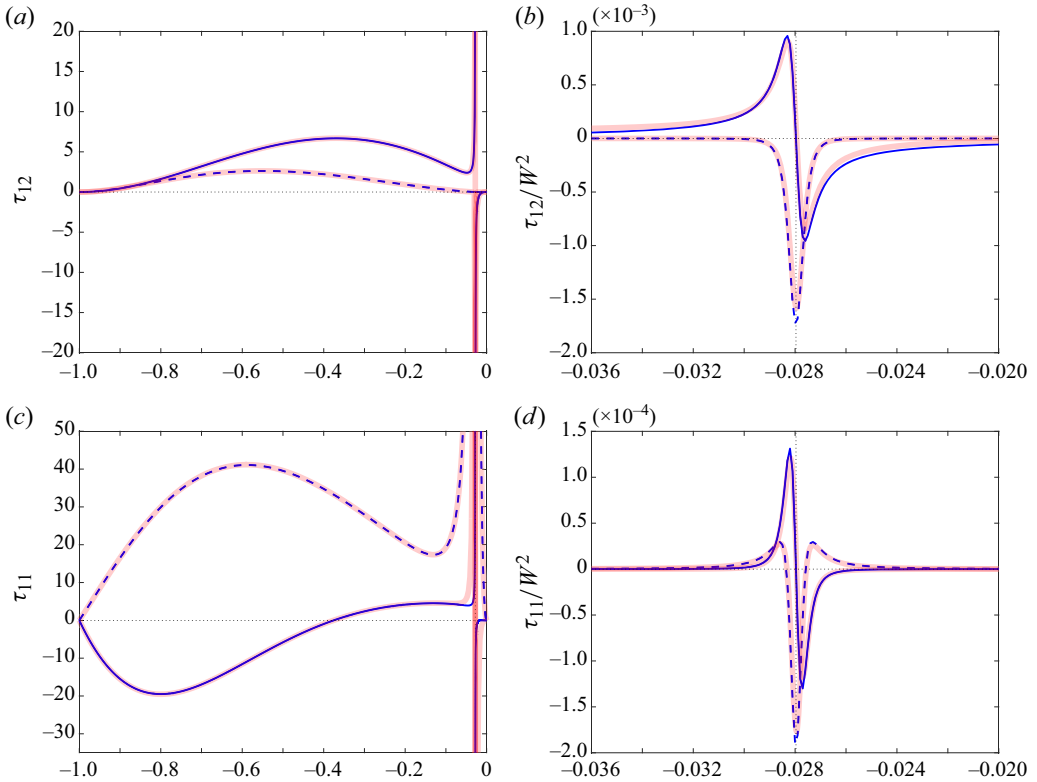


Figure 7. A comparison of the stresses τ_{12} (a,b) and τ_{11} (c,d) at $k = 1.1$, $\Lambda \approx 4.11$ computed numerically at $W = 32\,000$ (in blue in both (a,c) and (b,d)) with the outer solutions (a,c) and leading inner asymptotic solutions given by (4.24)–(4.25) (b,d – note the rescaling of both axes). The thick translucent red line is the numerical solution and the blue solid/dashed lines are the asymptotic approximation (as usual solid lines indicate real parts and dashed lines imaginary parts).

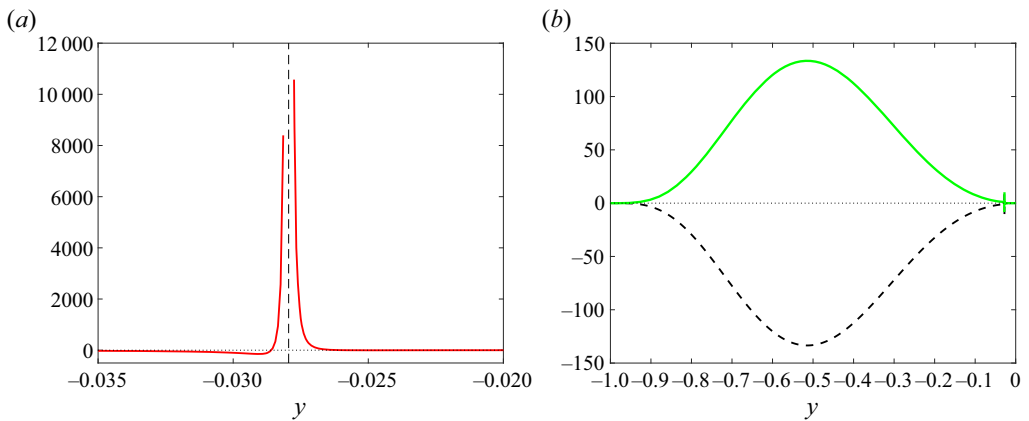


Figure 8. (a) $P(y)$ plotted against y for a solution ($k = 1.1$, $\Lambda = 4.520468$ and $c = 0.99921892$) matched across the critical layer using $\delta = 2 \times 10^{-4}$ (the position of the critical layer is shown as the vertical dashed black line). (b) $C(y)$ (green) and $D(y)$ (black dashed) plotted against y for the same matched solution. These figures show that the velocity field is driven locally by the critical layer whereas the polymer perturbation is driven globally by the gradual relaxation of the streamline distortion caused by the critical layer.

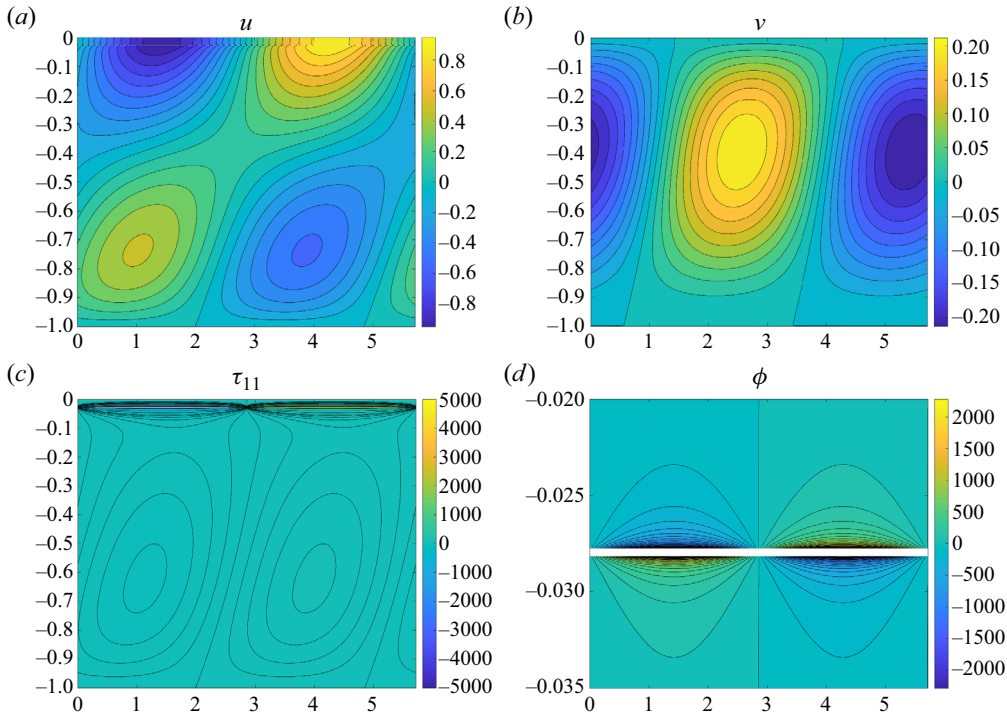


Figure 9. Contour plots of the outer solution corresponding to $k = 1.1$, $\Lambda = 4.520468$ and $c = 0.99921892$ matched across the critical layer using $\delta = 2 \times 10^{-4}$; (a) u , (b) v , (c) τ_{11} and (d) ϕ . The last plot focuses on the area around the critical layer (the white gap is just the excluded critical layer region $[y^* - \delta, y^* + \delta]$). For u and v , 20 equally spaced contour levels between the maximum and minimum values are used. For τ_{11} , the contour levels are $\pm[0, 10, 20, 30, 40, 50, 60, 100, 200, 1000, 2000, 5000]$ and 50 equally spaced levels are used for ϕ with all the contours essentially contained in the shown area around the critical layer.

$\phi(\pi/2, y)$ in figure 9 – and then blowing them apart – see $\phi(3\pi/2, y)$ in the same figure. This amplifies the base streamwise-normal stress field which in turn exerts a streamwise stress on the flow locally at the critical layer. The streamwise flow drives the cross-stream flow through continuity which then intensifies the critical layer closing the loop.

The one outstanding question is why the critical layer has to be so close to the centreline as $W \rightarrow \infty$. The asymptotic analysis above indicates that the shear at the critical layer needs to be $O(W^0)$ as $W \rightarrow \infty$ but can tell us nothing about the size of this $O(W^0)$ number relative to 1. In particular, given the complicated analysis, it is still not clear why the instability does not manifest in plane-Couette flow. To help answer this, we conduct some simple experiments in the next section.

6. Moving towards plane-Couette flow

The complexity of the matching analysis means it is difficult to discern the importance of U''_* or the size of U'_* (e.g. just look at the 3 constants that emerge in (4.64a–c)) despite our best efforts to keep them separated. So, here, we perform a series of numerical experiments exploring the effect of small changes which would bring the channel flow closer to plane-Couette flow (pCf). To keep things manageable, these experiments concentrate on studying how the lowest W point on the neutral curve, W_{min} , varies as the problem is changed slightly. Four experiments are undertaken in which this minimum point is tracked

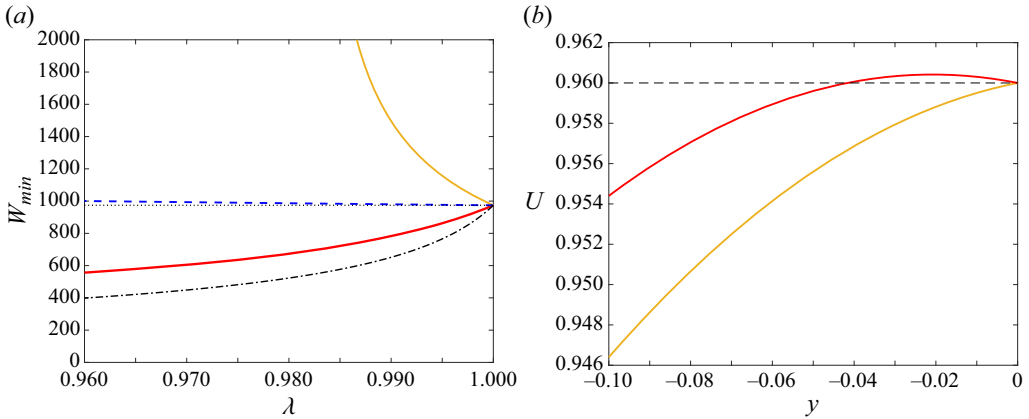


Figure 10. (a) The effect on W_{min} of decreasing λ from 1 to 0.96 (the common point on $\lambda = 1$ is the undisturbed channel flow value of $W_{min} = 974$): expt. 1 black dash-dot line; expt. 2 blue dashed line; expt. 3 yellow solid line; expt. 4 red solid line. (b) The base flow profile $U := \lambda(1 - y^2) + (1 - \lambda)y$ in expt. 3 (yellow line) and $U := \lambda(1 - y^2) + (1 - \lambda)(-y)$ in expt. 4 (red line) for $\lambda = 0.96$ showing how the former eliminates the zero-shear point whereas the latter moves it into the interior (notice only a small part of the flow domain is being shown close to the midplane. In all cases the problem is solved over $y \in [-1, 0]$).

as a homotopy parameter λ is reduced from 1, which is the channel flow problem studied above where $W_{min} = 974$. These are as follows (unless explicitly stated, the boundary conditions imposed on the disturbance field remain no slip at the wall and stress free at the midplane with the computation done across the half channel $y \in [-1, 0]$).

- (i) Expt. 1 explores the importance of U'' by (artificially) changing it while keeping U' fixed. Specifically $U'' := -2\lambda$ everywhere so U'' can be reduced without changing $U = 1 - y^2$ or $U' = -2y$ in the code.
- (ii) Expt. 2 explores the effect of changing the boundary condition at the midplane towards a solid boundary to mimic the monotonic increase in U across the domain of pCf. The boundary condition at $y = 0$ is set to $\lambda D^2 v + (1 - \lambda) Dv = 0$ so $\lambda = 1$ corresponds to the stress-free/symmetry conditions considered above and $\lambda = 0$ to a no-slip solid wall.
- (iii) Expt. 3 explores the effect of increasing the minimum shear across the domain $y \in [-1, 0]$. The base flow is set to $U := \lambda(1 - y^2) + (1 - \lambda)y$, which mixes in pCf in a way to gradually generate a (minimum) non-zero shear $= 1 - \lambda$ at the midplane.
- (iv) Expt. 4 explores the effect of moving the $U'' = 0$ point into the interior. The base flow is set to $U := \lambda(1 - y^2) + (1 - \lambda)(-y)$, which mixes in pCf in a way to move the zero-shear point at $y = -(1 - \lambda)/2\lambda$ away from the midplane and into the interior $y \in [-1, 0]$.

The results are shown in figure 10. The effect of changing the boundary conditions (blue dashed line) at the midplane is minimal and reducing U'' (black dash-dot line) is destabilising. The presence of vanishing shear, however, seems crucial: removing it (yellow line) quickly stabilises the instability whereas moving it from the midplane (red line) is destabilising. The plausible conclusion is that the instability needs an interior velocity maximum where the shear vanishes to nucleate. This is certainly absent in pCf. A very recent study by Yadav, Subramanian & Shankar (2024) (published after this paper

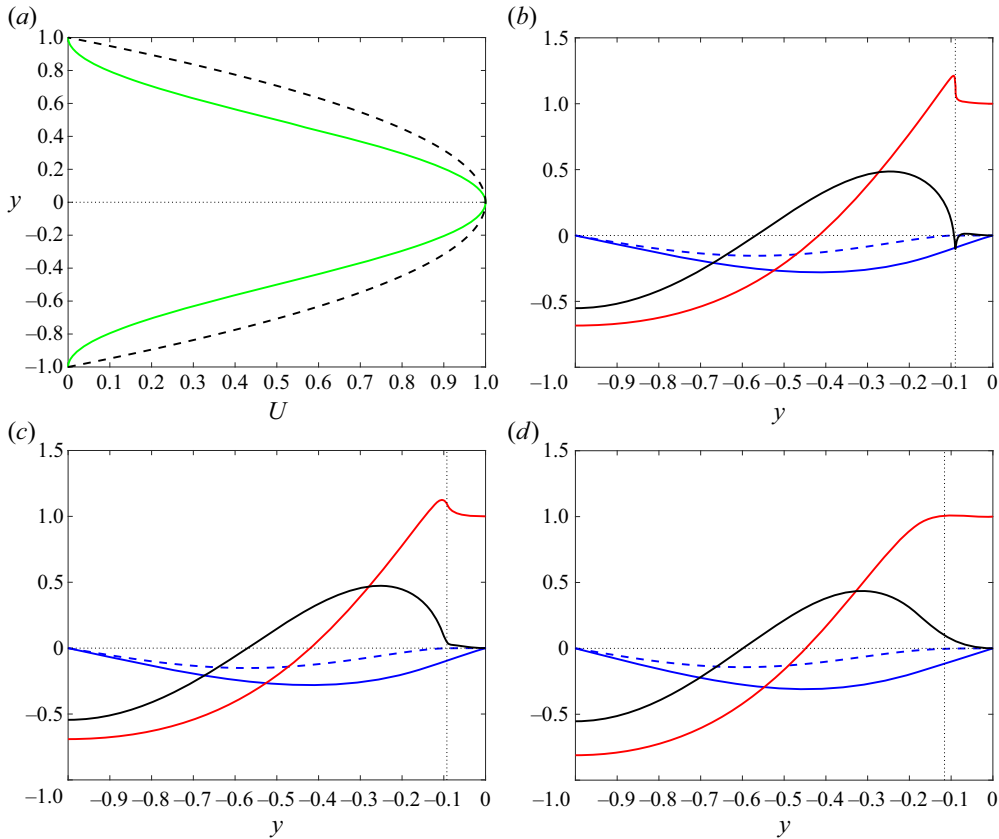


Figure 11. (a) Viscoelastic Kolmogorov base flow as given in (7.1) in green and channel flow $(1 - y^2)$ in dashed black. Neutral vKf eigenfunctions for $k = 1$ and $(W, \beta, c) = (2000, 0.99954, 0.98029)$ (b), $(W, \beta, c) = (200, 0.99530, 0.97897)$ (c) and $(W, \beta, c) = (20, 0.93972, 0.96726)$ (d). Here, v is blue (real/imaginary parts solid/dashed respectively), real part of Dv is red and imaginary part is black.

was submitted) of the viscoelastic plane-Couette-channel flow configuration has confirmed this as a necessary condition.

7. Viscoelastic Kolmogorov flow

Finally, the similarity of the base flow shape in viscoelastic Kolmogorov flow (vKf) to that in channel flow suggests that the centre-mode instability of Garg *et al.* (2018) and Khalid *et al.* (2021a,b) should be present in the results of Boffetta *et al.* (2005). It is straightforward to confirm this by renormalising the base flow in vKf to the form

$$U = \frac{1}{2}(1 + \cos \pi y) \quad (7.1)$$

(to most closely match $(1 - y^2)$ – see figure 11) and considering disturbances which are periodic over $y \in [-1, 1]$ and have the same symmetry (3.9) as the centre-mode instability about $y = 0$. These properties actually imply that the disturbance satisfies stress-free boundary conditions at $y = \pm 1$ and so all the numerical codes developed for channel flow can trivially be reapplied to vKf by just (i) changing $U(y)$ and (ii) imposing stress-free boundary conditions on the perturbation at $y = -1$.

A shooting code, which identifies the neutral curve at a given k and W after a guess for β and c (based on the channel flow solution), quickly identifies a neutral vKf eigenfunction for $(W, k) = (2000, 1)$ at $(\beta, c) = (0.99953538, 0.98029137)$. Reducing W to 200, keeping $k = 1$ and using the values of β and c found at $W = 2000$ as initial guesses, the neutral eigenfunction at $W = 200$ converges easily to $(\beta, c) = (0.99530288, 0.97896974)$. Repeating this procedure, reducing W from 200 to 20, again converges smoothly to $(\beta, c) = (0.93972375, 0.96725546)$; see figure 11. The similarity of the neutral eigenfunctions in figure 11 to those in channel flow (modulo the different boundary conditions at $y = -1$) is striking and suggests that Boffetta *et al.* (2005) were actually the first to see the centre-mode instability in rectilinear viscoelastic flow. To further support this conclusion, Berti *et al.* (2008); Berti & Boffetta (2010) see ‘arrowhead’ solutions when tracking their vKf instability to finite amplitude (e.g. see figures 7 and 8 in Berti & Boffetta 2010) just like those found in channel flow originating from the centre-mode instability; see Page *et al.* (2020), Buza *et al.* (2022a) and Beneitez *et al.* (2024).

Boffetta *et al.* (2005) considered lower values of β (e.g. 0.77 and 0.167 in their figure 2), disturbances with no specific symmetry (so both symmetric and antisymmetric ones as defined by (3.9)) and also long-wavelength perturbations compared with the forcing wavelength. While the observation made here is strictly for β close to 1 with disturbances having the symmetry (3.9) and same cross-stream wavelength as the forcing, a more complete investigation exploring these different aspects indicates that only the centre-mode instability mechanism operates in vKf (Lewy & Kerswell, unpublished).

8. Discussion

We first summarise the findings of the paper. The first part of these concern the $Re \rightarrow \infty$ asymptotics of the upper (§ 3.1) and lower branches (§ 3.2) of the centre-mode neutral curve in the Re – W plane for viscoelastic channel flow.

Along the upper branch

$$W \sim Re^{1/3}, \quad k \sim Re^{1/3}, \quad 1 - c \sim Re^{-2/3}, \quad (8.1a-c)$$

with numerical coefficients given in (3.11a–c) for $\beta = 0.9$. These scalings are equivalent to $Re \sim O(E^{-3/2})$, $k \sim O(E^{-1/2})$ and $1 - c \sim O(E)$ as the elasticity number $E := W/Re \rightarrow 0$, consistent with figure 11 in Khalid *et al.* (2021a).

Along the lower branch

$$W \sim Re, \quad k \sim \frac{1}{Re}, \quad c = O(1), \quad (8.2a-c)$$

with numerical coefficients computed for $\beta = 0.9, 0.98$ and 0.994 given in table 2. These lower branch scalings are apparent in figure 13 of Khalid *et al.* (2021a) (see also their figure 18).

The second part of the findings described in § 4 concern the inertialess limit of viscoelastic channel flow. By $\beta = 0.994$ as β increases, the lower branch has swung sufficiently clockwise in the Re – W plane to cross the $Re = 0$ axis (see figure 1). This reveals the existence of an inertialess ($Re = 0$) centre-mode instability and the $W \rightarrow \infty$ asymptotic problem in the ultra-dilute limit where $W(1 - \beta) = O(1)$ is then treated. A matched asymptotic analysis is performed in which a critical layer region is resolved sufficiently to extract matching conditions, (4.60)–(4.63), to connect up outer regions either side. Interestingly, the outer problem is fourth order as opposed to the

usual second-order problem for the Orr–Sommerfeld problem and so requires matching conditions all the way down to the third-order derivative in the cross-stream velocity. This leads to a particularly delicate matching procedure (§ 4.8) where the matching conditions need to be resolved to third order in the small matching parameter and quadruple precision is needed to make contact with numerical solutions. The completed analysis is successful in revealing, again unlike the Orr–Sommerfeld problem, that

$$1 - c = O(1), \quad \text{as } W \rightarrow \infty. \quad (8.3)$$

This $O(1)$ number can be deceptively small compared with 1 (e.g. 4.3×10^{-5} for $k = 0.1$ in table 3) but nevertheless remains finite as $W \rightarrow \infty$. That this has to be so is clear from the structure of the critical layer that is built around an $O(1)$ cross-stream velocity which has to be brought to zero at the midplane by an $O(1)$ outer region separating the critical layer from it. Quite why c has to be so close to 1 or equivalently why the critical layer sets up in a region of small shear is unclear (and unknowable from the asymptotic analysis). Some simple numerical experiments (§ 6) suggest that the lack of this small shear region is the likely reason the instability does not manifest in pCf.

The asymptotic analysis also clarifies that the instability mechanism (§ 5) is one in which the critical layer acts like a pair of ‘bellows’, periodically sucking the flow streamlines together and then blowing them apart (see figure 9). This amplifies the streamwise-normal polymer stress field which in turn exerts a streamwise stress on the flow locally at the critical layer. The streamwise flow drives the cross-stream flow by continuity, which then intensifies the critical layer, closing the loop.

Finally, in § 7, a connection is made between the centre-mode instability of channel flow and an earlier linear instability found in vKf by Boffetta *et al.* (2005). The fact that the instability in Kolmogorov flow was discovered at much lower Re and W to the extent it was viewed as a purely ‘elastic’ instability disguised its connection to the work of Garg *et al.* (2018). They worked at $Re = O(100)–O(1000)$ and $W \gtrsim 20$ in a very different geometry and so viewed their instability as ‘elasto-inertial’ in origin. It is clear in hindsight that the apparent difference in the regimes is more a function of the boundary conditions – a solid wall in the pipe versus periodicity in Kolmogorov flow – than any deeper dynamical difference as evident in the Newtonian versions of the respective problems ($Re_{crit} = O(10)$ for Kolmogorov flow while $Re_{crit} = 5772$ for channel flow).

The importance of the centre-mode instability for EIT, or indeed ET, is still an area of much current speculation (e.g. Datta *et al.* 2022; Dubief *et al.* 2023). While computations have confirmed that the instability leads to travelling wave solutions dubbed ‘arrowheads’ (Page *et al.* 2020; Buza *et al.* 2022a; Dubief *et al.* 2022), it remains unclear what these lead to via their own bifurcations. Recently Beneitez *et al.* (2024) have found that the arrowhead solutions coexist with EIT rather breaking down to it. The situation, however, is slightly clearer at $Re = 0$ (perhaps because the parameter space is one dimension less) where the (two-dimensional) arrowhead solution can become unstable to three-dimensional disturbances (Lellep *et al.* 2023). Very recent calculations using a large domain indicate that this instability can lead to a three-dimensional chaotic state (Lellep *et al.* 2024).

Further complicating the picture is the very recent emergence of another viscoelastic instability – dubbed ‘PDI’ for polymer diffusive instability – when polymer stress diffusion is present (Beneitez, Page & Kerswell 2023; Couchman *et al.* 2023; Lewy & Kerswell 2024). This is a ‘wall’ mode which also exists for all Re including 0 and any shear flow with a solid wall is susceptible. Beneitez *et al.* (2023) have already found that PDI can lead to a chaotic three-dimensional state in inertialess pCf using the FENE-P model.

Going forward, the challenge is to try to unpick which process of the current contenders – viscoelastic Tollmien Schlichting instability, the centre-mode instability or the PDI – triggers EIT and ET in what part of parameter space. This will assist in simulating EIT and ET and ultimately in manipulating those states as required for industrial applications.

Funding. The authors acknowledge EPSRC support for this work under grant EP/V027247/1 and the help of 3 referees in improving the presentation of the paper.

Declaration of interests. The authors report no conflict of interest.

Author ORCIDs.

 Rich R. Kerswell <https://orcid.org/0000-0001-5460-5337>;

 Jacob Page <https://orcid.org/0000-0002-4564-5086>.

Appendix A. Numerical methods for eigenvalue problem

Two complementary approaches were developed: a matrix formulation and a shooting technique.

A.1. Matrix

A generalised eigenvalue code was written to solve for all 6 variables ($u, v, p, \tau_{11}, \tau_{12}, \tau_{22}$) building in the symmetry of the unstable eigenfunction around the midline to improve efficiency. This was done by mapping the lower half of the channel $y \in [-1, 0]$ to the full Chebyshev domain $[-1, 1]$ so collocation points are concentrated at the wall and the centreline, and imposing symmetry boundary conditions at the centreline $y = 0$ which are $\partial u / \partial y = v = 0$ (no boundary conditions are imposed on p or the polymer stress τ anywhere). The fields are expanded using individual functions which incorporate these conditions, specifically

$$\begin{bmatrix} u \\ v \\ p \\ \tau_{11} \\ \tau_{12} \\ \tau_{22} \end{bmatrix} = e^{ik(x-ct)} \sum_{n=1}^N \begin{bmatrix} u_n \{T_{n+1}(2y+1) + \alpha_n T_n(2y+1) + (\alpha_n - 1)T_{n-1}(2y+1)\} \\ v_n \{T_{n+1}(2y+1) - T_{n-1}(2y+1)\} \\ p_n T_{n-1}(2y+1) \\ t_n T_{n-1}(2y+1) \\ r_n T_{n-1}(2y+1) \\ s_n T_{n-1}(2y+1) \end{bmatrix}, \quad (\text{A1})$$

where $T_n(z) := \cos(n \cos^{-1} z)$ is the n th Chebyshev polynomial, $(u_n, v_n, p_n, t_n, r_n, s_n) \in \mathbb{C}^6$ and $\alpha_n := -4n/(2n^2 - 2n + 1)$ so that $u(x, -1, t) = 0 = \partial u / \partial y(x, 0, t)$. A complementary inverse iteration code was also written which could take an eigenvalue from the generalised eigenvalue code and converge it at much higher resolution (e.g. table 5). This was important as the generalised eigenvalue problem is not well conditioned with increasing resolution; see the drift in the eigenvalue for $N \gtrsim 300$ in table 5 using *eig* in Matlab). This lack of conditioning gets worse near the neutral curve where an interior critical layer is present. Inverse iteration treats exactly the same matrices but is far better conditioned – there is no drift in table 5 even increasing N to 2000.

A.2. Shooting

Two shooting codes were also written based on different integrators. The first used the Runge–Kutta 4th order scheme (RK4) over a uniformly spaced grid (e.g. 50 000 points

	N	c_r	c_i
Eigenvalue code	100	0.999608051011	$8.2367630455 \times 10^{-5}$
	200	0.999608009040	$8.2751793055 \times 10^{-5}$
	300	0.999607999400	$8.2756092074 \times 10^{-5}$
	500	0.999608029589	$8.2771153285 \times 10^{-5}$
	1000	0.999608103554	$8.2644316360 \times 10^{-5}$
	2000	0.999607409613	$8.3760120903 \times 10^{-5}$
Inverse iteration	100	0.999608051169	$8.2367612043 \times 10^{-5}$
	200	0.999608007042	$8.2751523639 \times 10^{-5}$
	300	0.999608007115	$8.2751700997 \times 10^{-5}$
	500	0.999608007115	$8.2751701045 \times 10^{-5}$
	1000	0.999608007115	$8.2751701039 \times 10^{-5}$
	2000	0.999608007115	$8.2751701041 \times 10^{-5}$
Shooting	50 000	0.999608007115	$8.2751701040 \times 10^{-5}$
	ODE15s	0.999608007115	$8.2751701043 \times 10^{-5}$
	^a ODE15s	0.9996080071147197327	$8.27517010406007 \times 10^{-5}$

Table 5. Check of codes with the eigenvalue shown in figure 1 (inset) of Khalid *et al.* (2021*b*). The unstable centre mode at $\beta = 0.997$, $k = 0.75$ and $W = 2500$ is shown there with $c_r \approx 0.9996$ and $c_i \approx 8 \times 10^{-5}$. Two shooting codes were written based on different integrators. The first used the Runge–Kutta 4th order scheme (RK4) over a uniformly spaced grid (here 50 000 points across $[-1, 0]$) and the second used Matlab’s ODE15s with relative and absolute tolerances set at 3×10^{-14} (runtime 8 s on an Apple M1 processor).

^aODE15s is the quadruple precision extension using Advanpix with the quoted result showing the common digits using tolerances set at 10^{-20} (runtime 3 h 56 min) and 10^{-24} (runtime 21 h 29 min).

across $[-1, 0]$ in table 5) with inbuilt re-orthogonalisation of shooting solutions across the domain and a second used Matlab’s ODE15s with relative and absolute tolerances set at 3×10^{-14} which did not. For the solutions sought, re-orthogonalisation was not needed and so the latter, which was more efficient as it has locally adaptive stepping, was used for all subsequent calculations. The eigenvalue problem is fourth order so the usual shooting code approach takes a guess for the complex phase speed c and searches for the 2 unknown velocity boundary conditions at one wall which mean that the required boundary conditions at the other are obeyed. This can be readily adapted to search for the neutral curve directly by setting $c_i = 0$ and instead adjusting one (real) parameter of the problem. Here, we chose to vary $\Lambda = (1 - \beta)W$ keeping W fixed. This can be used to recreate the upper inset in figure 2 of Khalid *et al.* (2021*b*); see tables 2 and 3 for sample computations at $k = 0.1$ and $k = 1.1$.

REFERENCES

BENEITEZ, M., PAGE, J., DUBIEF, Y. & KERSWELL, R.R. 2024 Multistability of elasto-inertial two-dimensional channel flow. *J. Fluid Mech.* **981**, A30.
BENEITEZ, M., PAGE, J. & KERSWELL, R.R. 2023 Polymer diffusive instability leading to elastic turbulence in plane Couette flow. *Phys. Rev. Fluids* **8**, L101901.
BERTI, S., BISTAGNINO, A., BOFFETTA, G., CELANI, A. & MUSACCHIO, S. 2008 Two-dimensional elastic turbulence. *Phys. Rev. E* **77**, 055306.
BERTI, S. & BOFFETTA, G. 2010 Elastic waves and transition to elastic turbulence in a two-dimensional viscoelastic Kolmogorov flow. *Phys. Rev. E* **82**, 036314.
BOFFETTA, G., CELANI, A., MAZZINO, A., PULIAFITO, A. & VERGASSOLA, M. 2005 The viscoelastic Kolmogorov flow: eddy viscosity and linear instability. *J. Fluid Mech.* **523**, 161–170.

- BUZA, G., BENEITEZ, M., PAGE, J. & KERSWELL, R.R. 2022a Finite-amplitude elastic waves in viscoelastic channel flow from large to zero Reynolds number. *J. Fluid Mech.* **951**, A3.
- BUZA, G., PAGE, J. & KERSWELL, R.R. 2022b Weakly nonlinear analysis of the viscoelastic instability in channel flow for finite and vanishing Reynolds numbers. *J. Fluid Mech.* **940**, A11.
- CHAUDHARY, I., GARG, P., SUBRAMANIAN, G. & SHANKAR, V. 2021 Linear instability of viscoelastic pipe flow. *J. Fluid Mech.* **908**, A11.
- COUCHMAN, M.M.P., BENEITEZ, M., PAGE, J. & KERSWELL, R.R. 2023 Inertial enhancement of the polymer diffusive instability. *J. Fluid Mech.* **981**, A2.
- DATTA, S.S., *et al.* 2022 Perspectives on viscoelastic flow instabilities and elastic turbulence. *Phys. Rev. Fluids* **7** (8), 080701.
- DONG, M. & ZHANG, M. 2022 Asymptotic study of linear instability in a viscoelastic pipe flow. *J. Fluid Mech.* **935**, A28.
- DRAZIN, P.G. & REID, W.H. 1981 *Hydrodynamic Stability*. Cambridge University Press.
- DUBIEF, Y., PAGE, J., KERSWELL, R.R., TERRAPON, V.E. & STEINBERG, V. 2022 First coherent structure in elasto-inertial turbulence. *Phys. Rev. Fluids* **7** (7), 073301.
- DUBIEF, Y., TERRAPON, V.E. & HOF, B. 2023 Elasto-inertial turbulence. *Annu. Rev. Fluid Mech.* **55**, 675–705.
- GARG, P., CHAUDHARY, I., KHALID, M., SHANKAR, V. & SUBRAMANIAN, G. 2018 Viscoelastic pipe flow is linearly unstable. *Phys. Rev. Lett.* **121** (2), 024502.
- GROISMAN, A. & STEINBERG, V. 2000 Elastic turbulence in a polymer solution flow. *Nature* **405** (6782), 53–55.
- KHALID, M., CHAUDHARY, I., GARG, P., SHANKAR, V. & SUBRAMANIAN, G. 2021a The centre-mode instability of viscoelastic plane Poiseuille flow. *J. Fluid Mech.* **915**, A43.
- KHALID, M., SHANKAR, V. & SUBRAMANIAN, G. 2021b Continuous pathway between the elasto-inertial and elastic turbulent states in viscoelastic channel flow. *Phys. Rev. Lett.* **127** (13), 134502.
- LARSON, R.G., SHAQFEH, E.S.G. & MULLER, S.J. 1990 A purely elastic instability in Taylor–Couette flow. *J. Fluid Mech.* **218**, 573–600.
- LELLEP, M., LINKMANN, M. & MOROZOV, A. 2023 Linear instability analysis of purely elastic travelling waves in pressure-driven channel flows. *J. Fluid Mech.* **959**, R1.
- LELLEP, M., LINKMANN, M. & MOROZOV, A. 2024 Purely elastic turbulence in pressure-driven channel flows. *Proc. Natl Acad. Sci.* **121**, e2318851121.
- LEWY, T. & KERSWELL, R.R. 2024 The polymer diffusive instability in highly concentrated polymer fluids. *J. Non-Newtonian Fluid Mech.* **326**, 105212.
- MOROZOV, A. 2022 Coherent structures in plane channel flow of dilute polymer solutions with vanishing inertia. *Phys. Rev. Lett.* **129** (1), 017801.
- PAGE, J., DUBIEF, Y. & KERSWELL, R.R. 2020 Exact traveling wave solutions in viscoelastic channel flow. *Phys. Rev. Lett.* **125** (15), 154501.
- RALLISON, J.M. & HINCH, E.J. 1995 Instability of a high-speed submerged elastic jet. *J. Fluid Mech.* **228**, 311–324.
- SAMANTA, D., DUBIEF, Y., HOLZNER, M., SCHÄFER, C., MOROZOV, A.N., WAGNER, C. & HOF, B. 2013 Elasto-inertial turbulence. *Proc. Natl Acad. Sci.* **110** (26), 10557–10562.
- SHAQFEH, E.S.G. 1996 Purely elastic instabilities in viscometric flows. *Annu. Rev. Fluid Mech.* **28**, 129–185.
- SHEKAR, A., MCMULLEN, R.M., MCKEON, B.J. & GRAHAM, M.D. 2021 Tollmien–Schlichting route to elastoinertial turbulence in channel flow. *Phys. Rev. Fluids* **6**, 093301.
- SHEKAR, A., MCMULLEN, R.M., WANG, S.-N., MCKEON, B.J. & GRAHAM, M.D. 2019 Critical-layer structures and mechanisms in elastoinertial turbulence. *Phys. Rev. Lett.* **122**, 124503.
- SID, S., TERRAPON, V.E. & DUBIEF, Y. 2018 Two-dimensional dynamics of elasto-inertial turbulence and its role in polymer drag reduction. *Phys. Rev. Fluids* **3** (1), 011301.
- WAN, D., SUN, G. & ZHANG, M. 2021 Subcritical and supercritical bifurcations in axisymmetric viscoelastic pipe flows. *J. Fluid Mech.* **929**, A16.
- YADAV, S.K., SUBRAMANIAN, G. & SHANKAR, V. 2024 Elastic instability in a family of rectilinear viscoelastic channel flows devoid of centerline symmetry. *Phys. Rev. Fluids* **9**, 013301.